CE-317 STRUCTURAL ANALYSIS I DEPARTMENT OF CIVIL ENGINEERING Midsem; September 16, 2019, 11am-1pm

Problems carry equal weightage

Problem 1 YOU MUST USE ONLY SLOPE DEFLECTION METHOD

Refer Fig. 1. For the frame having an internal hinge at *C*, and loaded as shown, **determine** the rotation at *B* (i.e., θ_B) and reactions (i.e., forces and moment) at the support *D*. Use $EI = 4 \times 10^{13}$ N.mm² for all members, and w = 6 kN/m.



Problem 2 YOU MUST USE ONLY MOMENT DISTRIBUTION METHOD

Refer Fig. 2. For the frame loaded as shown, **determine all support reactions (i.e., forces and moments)** based on two iterations of moment distribution. Use $EI = 4 \times 10^{13}$ N.mm² for all members, and P = 50 kN.

Problem 3 YOU MUST USE ONLY STIFFNESS MATRIX METHOD

Refer Fig. 3. Determine all member forces for the truss loaded as shown. Use the joint and member notation as shown. Use $EA = 2 \times 10^8$ N for all members, and P = 50 kN.



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TA. 71 D.o. sway = 1. Deformed chords PAB = 0 YBC=-4CD=-A L=20M MBA = 3EI OB + WLZ $M_{BC} = 3 \underbrace{EI}_{L} \left(\Theta_{B} - \Psi_{BC} \right) = 3 \underbrace{EI}_{L} \left(\Theta_{B} + \underbrace{A}_{L} \right)$ $M_{DC} = 3EI(-\psi_{DC}) = -3EIA$ $M_{BA} + M_{BC} = 0 = 6 \stackrel{\text{EI}}{=} 0_B + 3 \stackrel{\text{EI}}{=} \stackrel{\text{A}}{=} + \frac{WL^2}{8} \longrightarrow O$ $\int O_{L} = V_{C+} \rightarrow M_{BC} = M_{DC} \rightarrow O_{B} = -2A \rightarrow \textcircled{}$ $\mathcal{O}_{\mathcal{O}} \xrightarrow{\mathcal{O}} \stackrel{\mathcal{O}}{\longrightarrow} \stackrel{\mathcal{O}}{=} = \frac{WL^3}{EI} \frac{1}{72}, \quad \mathcal{O}_{\mathcal{B}} = -\frac{WL^3}{EI} \frac{1}{36} = 0.0333 \, \text{rad}.$ $M_{DC} = -\frac{WL}{24} = -100 \text{ RNM}$ FE DENA $D_y = \frac{M_{DC}}{1} = -\frac{WL}{24} = -5RN$ EN SEN $D_x = \frac{1}{16} \left(-\frac{WL^2}{2} - M_{DC} + D_{y.52} \right) = -85 \text{KN}$

$$\begin{array}{c} \frac{P2}{P2} & A_{CD} = \frac{A}{5ing} = \frac{5}{4}A \qquad (2)$$

$$A_{CD} = \frac{A}{5ing} = \frac{5}{4}A \qquad (2)$$

$$A_{BC} = \frac{A}{tana} = \frac{3}{4}A \qquad (2)$$

$$N_{0} = \frac{A}{tana} = \frac{3}{4}A \qquad (2)$$

$$R = \frac{A}{tana} = \frac{3}{4}A \qquad (2)$$

$$R = \frac{A}{tana} = \frac{1}{4}A \qquad (3)$$

$$R = \frac{A}{tana} = \frac{1}{4}$$

$$\begin{split} \mathsf{M}_{\mathsf{BA}} &= 40.46 \ , \text{ and } \text{ sim, } \text{ larky for } \mathsf{M}_{\mathsf{BC}}, \mathsf{M}_{\mathsf{CB}}, \mathsf{M}_{\mathsf{CD}}, \overset{\textcircled{}}{\mathsf{M}}_{\mathsf{CD}}, \overset{\r{}}{\mathsf{M}}_{\mathsf{CD}}, \overset{\r{}}{\mathsf{M}}_{\mathsf{M}}, \overset{\r{}$$

 $F_{00} = \frac{EA}{1\sqrt{2}} \left[\frac{1}{12} - \frac{1}{12} \right] \left[\frac{8.0880}{0} \right] = \frac{1}{14} \left[\frac{1}{12} + \frac{1}{12} \right] \left[-\frac{30.9644}{14} \right] = \frac{1}{14} \left[\frac{1}{12} + \frac{1}{12} \right] \left[\frac{1}{12} + \frac{1}{$ $F_{GG} = \frac{1}{12} \left\{ \left(\frac{1}{12} + \frac{1}{12} \right) \left(\frac{1}{-30 \cdot 9645} \right)^2 = -15 \cdot 1822 = F_{G} \right\}$ = -11-4382 = FI FBG=0 (no jt displacement) = FB $\overline{GG} = \begin{bmatrix} 0 \\ 1 \end{bmatrix} \begin{bmatrix} 0 \\ -30.9644 \end{bmatrix} = -30.9644 = \boxed{14}$ $F_{00} = (1 \ 0) [8.088] = 8.088 = F_{5}$





@ support D

$M_{PC} = \underbrace{3EI}_{L} \Psi = -\underbrace{3EI}_{L} \times \underbrace{\omegal^{3}}_{72EI} = -\underbrace{\omegal^{2}}_{24} \mathcal{L}$

$V_D = M_{DC} = \omega + 1 = Dy$





$D_x = 85 \text{kN} \leftarrow$







No internal forces



 $M_{CD} = M_{DC} = \frac{6ET}{5} \Psi = 100$

 $M_{BC} = -M_{CB} = -\frac{6EI}{6} \times \frac{\psi}{2} = -\frac{6EI}{5} \psi \times \frac{5}{12}$ $= -100 \times \frac{5}{12} = -\frac{125}{3}$

 $M_{BA} = M_{AB} = \frac{GEI}{10} \times \frac{2}{5} \psi = \frac{GEI}{5} \psi \times \frac{1}{5} = 20$









