

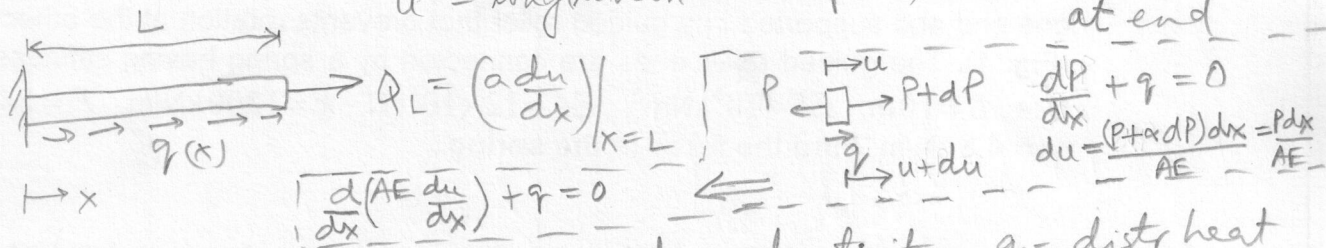
2nd Order BVP

$$-\frac{d}{dx} \left(a \frac{du}{dx} \right) + cu - q = 0, \quad 0 < x < L, \quad a = a(x), \quad c = c(x), \quad q = q(x)$$

$$u(0) = u_0, \quad a \left(\frac{du}{dx} \right) \Big|_{x=L} = Q_L$$

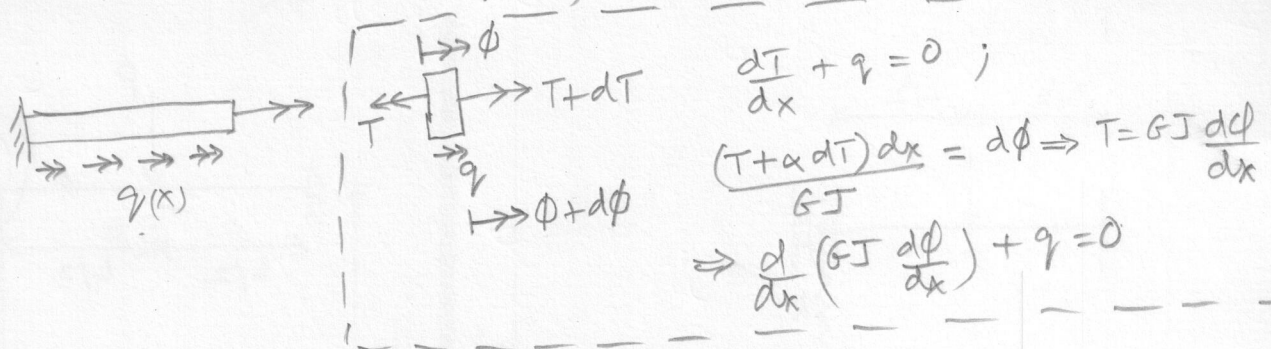
Examples ($c(x) = 0$):

Axial bar: $a = EA$, $q =$ distributed traction,
 $u =$ longitudinal displ, $Q_L =$ axial force at end

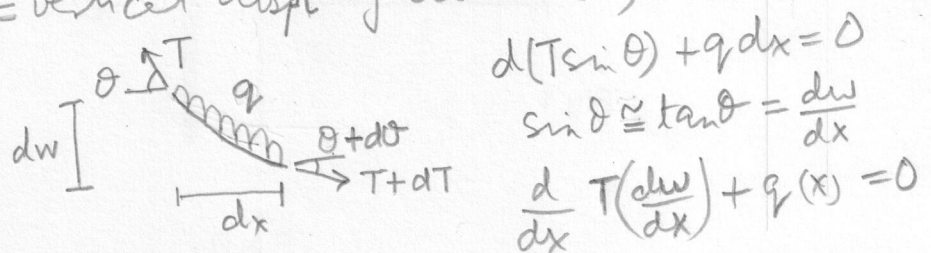


1-D Conduction: $a =$ thermal conductivity, $q =$ distr heat source
 $u =$ Temp, $Q_L =$ Heat out at $x = L$.

Torsion: $a = GJ(x)$, $q =$ distributed torque,
 $u =$ rotation $\phi(x)$, $Q_L =$ end torque T_L .



Cable: $a = T$ (cable tension), $q =$ distributed load,
 $u =$ vertical displ of cable $w(x)$, $Q_L =$



Flow thru pipes (Darcy): $a = \frac{\pi D^4}{128 \mu}$, $u \equiv$ hydrostatic pressure
 $q =$ flow source, $Q_L =$ flow rate.

Laminar flow thru channel: $a =$ viscosity, $u =$ velocity, $q =$ pressure gradient
 incompressible under constant pressure gradient $Q_L =$ axial stress

Flow thru porous media: a = coeff of permeability; u = fluid head,
 q = fluid flux, Q_L = flow seepage. (2)

Electrostatics: a = dielectric const, u = electrostatic potential,
 q = charge density, Q_L = electric flux.

Approx solution

$$u \approx U_N = \sum_{j=1}^N C_j \phi_j(x) + \phi_0(x)$$

$\phi_0(0) = u_0$, $\phi_i(0) = 0$, i.e. ϕ_0 used to satisfy
non-homogenous essential BC's.

$\phi_0(x)$, $\phi_i(x)$ chosen, usually polynomials.

In general, if you chose U_N to satisfy all
BC's (say for $u_0 = 1$, $Q_L = 0$, $a(x) = x$, $C(x) = 1$,
 $q(x) = 0$, $L = 1$, choose $\phi_0(x) = 1$, $\phi_1(x) = x^2 - 2x$,
 $\phi_2(x) = x^3 - 3x$, $N = 2$, $U_N = 1 + C_1(x^2 - 2x)$
 $+ C_2(x^3 - 3x)$)

subst U_N in BVP, get,

$$(2C_1 + 3C_2 + 1) + (-4C_1 - 2C_2 - 3C_2)x + (-9C_2 + C_1)x^2 + (C_2)x^3 = 0$$

$\Rightarrow$ each $() = 0 \Rightarrow$ inconsistent set of equations
for C_1, C_2 .

So we need a weighted-integral or weak-form
weak formulation of the BVP.

Weak Form (Weighted Integral/Residual) ③

$$\int_0^L W \left(-\frac{d}{dx} \left(a \frac{du}{dx} \right) + cu - q \right) dx = 0 \rightarrow \textcircled{1} \quad W = W(x)$$

$$\int_0^L \left(a \frac{dW}{dx} \frac{du}{dx} + cWu - Wq \right) dx - W a \frac{du}{dx} \Big|_0^L = 0 \rightarrow \textcircled{2}$$

→ Eq ① yields non-symmetric form. requires u to be twice differentiable, but not so for weight fn W . So choosing $u = u_N$ that satisfies all BC's, writing ① N times using $W = W_1, \dots, W_N(x)$ we can get N eqns for $C_1, \dots, C_N$. This is weighted-residual method.

wt-Residual:
Galerkin $\rightarrow W_i = \phi_i$
Collocation $\rightarrow W_i = \delta(x - x_i)$

→ Eq ② requires u to be once differentiable, i.e. weaker continuity requirement on u . Also it often results in a symmetric form ($W \approx u$ in first two terms under integral), yielding symmetric algebraic eqns. Also natural BC (i.e. $a \frac{du}{dx}$) included in ② so u need only satisfy geometric BC's. ② is the WEAK FORM. Note that such trading of differentiation from u to W should be avoided if it yields physically meaningless boundary terms. Weak form imposes continuity/differentiability requirements on W .

→ Primary variables ^(PV) $\rightarrow w$ and its derivatives in bndry terms, written in terms of u (i.e. $w \rightarrow u$), i.e., u in present problem. Specification of primary variables at bndry's constitute Essential/Kinematic/Geometric BC's ^(EBC), i.e. $u(0) = u_0$ in present problem.

→ Secondary variables ^(SV) $\rightarrow$ coeffs on w and its derivatives in bndry term, i.e. $a \frac{du}{dx}$ in present case. Specification of SV at bndry's constitutes Natural/Force BC's ^(NBC), i.e. $a \frac{du}{dx} \Big|_{x=L} = Q_L$ in present case.

→ $2m^{\text{th}}$ order BVP will have m pairs of (PV, SV). Each PV has associated SV. (4)

→ Apart from differentiability/continuity of w as dictated by weak form (2), w is required to satisfy homogeneous form of EBC's, i.e. $w(0) = 0$ in present case.

Thus weak form (2) becomes,

$$\int_0^L \left(a \frac{dw}{dx} \frac{du}{dx} + cwu - wq \right) dx - W(L) Q_L = 0 \quad \text{WEAK FORM of 2nd Order BVP} \quad \text{--- (2)*}$$

→ Weak form exists for all linear/nonlinear BVP's of order ≥ 2 .
When BVP is linear and even ordered, weak form is symmetric and Bilinear in w, u .

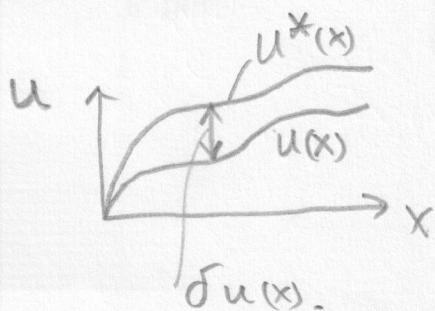
$$B(w, u) \triangleq \int_0^L \left(a \frac{dw}{dx} \frac{du}{dx} + cwu \right) dx \quad \text{--- Bilinear, symmetric in } w, u.$$

$$L(w) \triangleq \int_0^L wq dx + W(L) Q_L \quad \text{--- Linear in } w.$$

$$B(w, u) - L(w) = 0 \quad \text{--- (3) WEAK FORM. where } B(w, u) \text{ is sym \& bilinear, } L(w) \text{ is linear.}$$

$$\Rightarrow B(\epsilon w, u) - L(\epsilon w) = 0$$

Consider $\underbrace{u^*(x)}_{\text{varied soln}} = \underbrace{u(x)}_{\text{actual soln}} + \underbrace{\delta u(x)}_{\text{variation or small arbitrary change in } u(x), \text{ where } \delta u(x) \text{ satisfies homogeneous EBC's of } u. \text{ Thus we can write } \delta u(x) \equiv \epsilon w(x)}$



⑤

→ $\delta u(x) \neq du(x)$
 virtual change in actual soln $u(x)$ at a station x actual/real change in soln $u(x)$ across stations dx apart.

→ However, laws of differential calculus apply to $\delta u(x)$. Note $\delta x = 0$ (no variation of indep variable).

Symmetric Bilinear WEAK FORM ③ becomes,

$$B(\delta u, u) - L(\delta u) = 0$$

$$B(\delta u, u) = \int_0^L \left(a \frac{d\delta u}{dx} \frac{du}{dx} + c \delta u \cdot u \right) dx = \frac{1}{2} \delta \left[a \left(\frac{du}{dx} \right)^2 + cu^2 \right] dx$$

$$= \frac{1}{2} \delta B(u, u)$$

$$\Rightarrow \delta \left[\frac{1}{2} B(u, u) - L(u) \right] = \delta I(u) = 0 \quad \left(\text{used } L(\delta u) = \delta L(u) \right)$$

(used $L(\delta u) = \delta L(u)$ because L is linear.)

Thus for symmetric bilinear $B(u, u)$ and linear $L(u)$, WEAK FORM becomes

$\delta I(u) = 0$

$I(u) = \frac{1}{2} B(u, u) - L(u)$

WEAK FORM Extremum Condtn. Quadratic functional (in u)

In Sol Mech $\equiv$ Min PE, $I \equiv PE$.

$$L(\delta u) = \int_0^L \delta u \cdot q \, dx + \delta u(L) Q_L$$

$$= \delta \left[\int_0^L u \cdot q \, dx + u(L) Q_L \right]$$

$$= \delta [L(u)]$$

where we used $\delta(u \cdot q) = \delta u \cdot q$ because $\delta q = 0$
 $\delta(u(L) Q_L) = \delta u(L) Q_L$ because $\delta Q_L = 0$
 where $q(x), Q_L$ are prescribed.

Thus in Sol Mech, WEAK FORM $\equiv$ MIN. PE.

→ FEM doesn't require quad functional $I(u)$. Can directly work with WEAK FORM which is more general than $I(u)$.

$$I(u) = \underbrace{\frac{1}{2} \int_0^L \left[a \left(\frac{du}{dx} \right)^2 + cu^2 \right] dx}_\text{S.E} - \underbrace{\left[\int_0^L u \cdot q \, dx + u(L) Q_L \right]}_{\text{work done by } q + \text{work done by } Q_L}$$

$\frac{1}{2} \int_0^L \sigma_x \epsilon_x \, dV$
 $= \frac{1}{2} \int_0^L E \epsilon_x^2 A \, dx$
 $= \frac{1}{2} \int_0^L a \left(\frac{du}{dx} \right)^2 dx$
 $\equiv$ S.E due to discrete springs

(Variational) Methods of Approximation.

(6)

(I) Rayleigh - Ritz method — uses WEAK FORM.

$$U_N = \sum_{j=1}^N c_j \phi_j + \phi_0, \quad \underline{W = \phi_i}, \quad i=1, \dots, N.$$

$$\underbrace{B(w, u)}_{\text{bilinear}} = \underbrace{L(w)}_{\text{linear.}}$$

$$B(\phi_i, \sum_{j=1}^N c_j \phi_j + \phi_0) = L(\phi_i) \quad i=1, \dots, N$$

$$\because B \text{ bilinear} \Rightarrow \sum_{j=1}^N B(\phi_i, \phi_j) c_j = \underbrace{L(\phi_i) - B(\phi_i, \phi_0)}_{F_i}$$

$$\sum_{j=1}^N B_{ij} c_j = F_i \quad \rightarrow \textcircled{1}$$

$\rightarrow$ Ritz Eqs for c_j

Additionally, if B is also symmetric then $\exists I(u)$ and WEAK FORM is $\delta I(u) = 0$

$$\Rightarrow \frac{\delta I}{\delta c_i} = 0, \quad i=1, \dots, N \quad \rightarrow \text{Ritz Eqs for } c_j \rightarrow \textcircled{1}^*$$

$\textcircled{1} = \textcircled{1}^*$ but $\textcircled{1}$ more general as it requires only bilinearity of $B(w, u)$

$\rightarrow \phi_0$ required only to satisfy non-homo EBC's, i.e.,

$$u(x_0) = U_N(x_0) = \sum_{j=1}^N c_j \phi_j(x_0) + \phi_0(x_0) \quad \begin{matrix} \downarrow \neq 0 & \downarrow = u_0 \end{matrix}$$

$\rightarrow \phi_0$ should be sufficiently differentiable as required by $B(\phi_i, \phi_0)$

$\rightarrow \phi_j$ should be sufficiently differentiable as required by $B(\phi_i, \phi_j)$

$\rightarrow \phi_i$ { should satisfy homogenous EBC's.
 should be linearly independent and should yield indep rows/cols in $[B_{ij}]$
 should form complete set. }

Ex

$$-\frac{d^2 u}{dx^2} - u + x^2 = 0, \quad 0 < x < 1$$

(7)

$$u(0) = u(1) = 0$$

$$\phi_0 = 0, \quad \phi_i = x^i(1-x)$$

$$\begin{aligned} B_{ij} &= \int_0^1 \left(\frac{d\phi_i}{dx} \frac{d\phi_j}{dx} - \phi_i \phi_j \right) dx \\ &= \int_0^1 \left\{ [i x^{i-1}(1-x) - x^i] [j x^{j-1}(1-x) - x^j] - x^{i+j}(1-x)^2 \right\} dx \\ &= \frac{2ij}{(i+j)[(i+j)^2-1]} - \frac{2}{(i+j+1)(i+j+2)(i+j+3)} \end{aligned}$$

$$F_i = \int_0^1 -\phi_i x^2 dx = - \int_0^1 x^2 x^i(1-x) dx = \frac{1}{(3+i)(4+i)}$$

$$N=2 \rightarrow \frac{1}{420} \begin{bmatrix} 126 & 63 \\ 63 & 52 \end{bmatrix} \begin{Bmatrix} c_1 \\ c_2 \end{Bmatrix} = -\frac{1}{60} \begin{Bmatrix} 3 \\ 2 \end{Bmatrix}$$

$$c_1 = -\frac{10}{123}, \quad c_2 = -\frac{21}{123}, \quad U_{N=2} = -\frac{1}{123} (10x + 11x^2 - 21x^3)$$

$$\text{Exact soln: } u(x) = \frac{\sin x + 2 \sin(1-x)}{\sin 1} + x^2 - 2$$

$$U_{N=2} \approx u, \quad U_{N=3} \approx u \rightarrow \text{ie overlaps in plot, but not exactly equal}$$

$$\text{If BC is } u(0)=0, \quad \left. \frac{du}{dx} \right|_{x=1} = 1$$

$$\phi(0)=0, \quad \phi_i = x^i$$

$$B_{ij} = \int_0^1 (ij x^{i+j-2} - x^{i+j}) dx$$

$$F_i = - \int_0^1 x^{i+2} dx + [i] \quad \text{and proceed.}$$

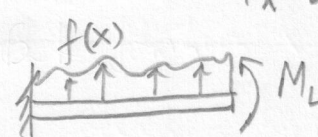
$$\text{Here too } U_{N=2} \approx u, \quad U_{N=3} \approx u \xrightarrow{\text{almost same}} u = \frac{2 \cos(1-x) - \sin x}{\sin 1} + x^2 - 2$$

Ex

$$\frac{d^2}{dx^2} \left[b(x) \frac{d^2 w}{dx^2} \right] - f(x) = 0, \quad 0 < x < L$$

(8)

$$w(0) = \frac{dw}{dx} \Big|_{x=0} = 0, \quad \left(b \frac{d^2 w}{dx^2} \right)_{x=L} = M_L, \quad \left[\frac{d}{dx} \left(b \frac{d^2 w}{dx^2} \right) \right]_{x=L} = V_L = 0$$

$w \uparrow$  Euler Bernoulli cantilever beam with $f(x)$, M_L applied
 $b(x) = EI(x)$

$$\int_0^L \left(b(x) \frac{d^2 v}{dx^2} \frac{d^2 w}{dx^2} - f v \right) dx + v \frac{d}{dx} \left(b \frac{d^2 w}{dx^2} \right) \Big|_0^L - \frac{dv}{dx} b \frac{d^2 w}{dx^2} \Big|_0^L = 0$$

$\left[w, \underbrace{\frac{d}{dx} \left(b \frac{d^2 w}{dx^2} \right)}_M \right]$ and $\left[\underbrace{\frac{dw}{dx}}_V, \underbrace{b \frac{d^2 w}{dx^2}}_M \right]$ are the two (PV, SV) pairs.
 $V = SF$

$$B(v, w) = \int_0^L b(x) v'' w'' dx, \quad L(v) = - \int_0^L f(x) v dx + v'(L) M_L$$

$B(v, w) - L(v) = 0 \rightarrow$ WEAK FORM where we used $v(0) = v'(0) = V_L = 0$.

$$I(w) = \underbrace{\frac{1}{2} \int_0^L b (w'')^2 dx}_{SE} - \underbrace{\left(\int_0^L f w dx \right)}_{\text{work done by } f(x)} + \underbrace{w'(L) M_L}_{\text{work done by } M_L}$$

$$\left(\frac{1}{2} \int \sigma_x \epsilon_x dV = \frac{1}{2} \int E \epsilon_x^2 dV = \frac{1}{2} \int_0^L \int_A E (y w'')^2 dA dx = \frac{1}{2} \int_0^L EI (w'')^2 dx \right)$$

$\delta I(w) = 0 \rightarrow$ Also WEAK FORM

Rayleigh Ritz solution: for $b(x) = EI = \text{const}$, $f(x) = f_0 = \text{const}$.
 $\phi_i(x) = x^{i+1}$ (twice differentiable, satisfies homogeneous form of EBC's)
 $\phi_0(x) = 0$ (EBC's homogeneous, ie no inhomogeneous term).

$$B_{ij} = \int_0^L EI (i+1)(j+1)(j) x^{i+j-2} dx = EI \frac{(i+1)(j+1)L^{i+j-1}}{i+j-1} = B_{ji}$$

$$F_i = \int_0^L f_0 x^{i+1} dx + (i+1)L^i M_L = f_0 \frac{L^{i+2}}{i+2} + (i+1)L^i M_L$$

For $N=2$ (two term soln),
$$\begin{bmatrix} B_{11} & B_{12} \\ B_{21} & B_{22} \end{bmatrix} \begin{Bmatrix} C_1 \\ C_2 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \end{Bmatrix}$$

(9)

gives
$$W_{N=2} = \underbrace{\frac{5f_0 L^2 + 12M_0 L}{24EI}}_{C_1} x^2 + \underbrace{\left(\frac{-f_0 L}{12EI}\right)}_{C_2} x^3 \approx W_{\text{exact}}$$

For $N=3$,
$$\begin{bmatrix} B_{11} & B_{12} & B_{13} \\ B_{21} & B_{22} & B_{23} \\ \text{symm} & & B_{33} \end{bmatrix} \begin{Bmatrix} C_1 \\ C_2 \\ C_3 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \\ F_3 \end{Bmatrix}$$

gives
$$W_{N=3} = \sum_{i=1}^3 C_i \phi_i = \frac{f_0 x^2 (6L^2 - 4Lx + x^2)}{24EI} + \frac{M_0 x^2}{2EI}$$

$$= W_{\text{exact}}$$

 (exactly true).

$\Rightarrow$ For $N=4$, we will get $C_i = 0, i > 3$.

(II) Weighted Residual Methods.

$\phi_j \rightarrow$ differentiable as much as (original) BVP requires
 satisfies homogenous form of ALL BC's.

$\phi_0 \rightarrow$ satisfies ALL BC's.

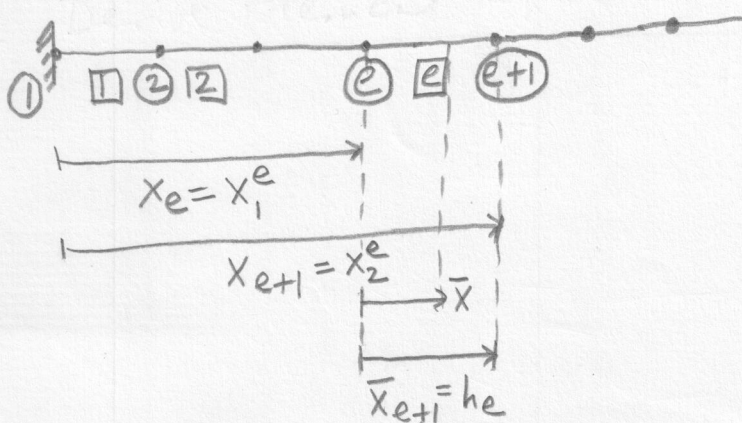
Galerkin method: $w_j = \phi_j$

Collocation method: $w_j = \delta(x - x_j)$

Approx fns ϕ_i, ϕ_0 , difficult to generate in a systematic way for complex geometries. Only continuity, completeness, EBC satisfaction, Linear independence requirements prescribed but their systematic generation for complex geometries is not prescribed. Hence, ^{traditional variational methods such as} RR, Galerkin, etc methods, where ϕ_i, ϕ_0 prescribed over entire domain, is not competitive. In FEM the complex geometry doesn't alter the approx fns.

- Unlike traditional variational methods where ϕ_0, ϕ_i must satisfy certain BC's (eg ϕ_0 satisfies all EBC's, ϕ_i satisfy all homogeneous form of EBC's in Rk-method) such that U_N satisfies EBC's (in RR) or ALL BC's (in weighted residual), in FEM the approx/shape fns have no BC satisfaction requirements, since interelement continuity is imposed and BC's are imposed at the end as in stiffness method.

Step 1 Discretize domain (number nodes, elements, generate geometric properties of elements and coords of nodes).



Linear 2-noded elements.

Step 2 Derive Element Equations. (11) Relation between PV's & SV's

Substep (i) Weak form or Weighted Residual

↳ (we do this, ie RR-method)

2-noded linear Lagrange, C^0

$$Q_1^e = -\left(\frac{adu}{dx}\right)_{x_1^e}, \quad Q_2^e = \left(\frac{adu}{dx}\right)_{x_2^e}$$

3-noded Quadratic Lagrange, C^0

$$Q_1^e = -\left(\frac{adu}{dx}\right)_{x_1^e}, \quad Q_3^e = \left(\frac{adu}{dx}\right)_{x_3^e}$$

Here $n \equiv N$ in traditional Variational Methods. We changed notation $\therefore$ later we encounter N_i shape function

$$U^e = \sum_{j=1}^{n=\text{nos. nodes per element}} u_j^e \phi_j^e(x), \quad x_1 < x < x_n \text{ (ie } x \text{ lies within element } e)$$

- weak form requires ϕ_j^e at least linear. Weighted residual requires ϕ_j^e at least quadratic.

Weak form is,

$$\int_{x_1^e}^{x_2^e} \left(a \frac{dw}{dx} \frac{du}{dx} + cwu - wq \right) dx - w(x_1^e) Q_1^e - w(x_n^e) Q_n^e$$

$$Q_1^e = -\left(\frac{adu}{dx}\right)_{x_1^e}, \quad Q_n^e = \left(\frac{adu}{dx}\right)_{x_n^e}, \quad \text{i.e., both} \rightarrow \text{+ve}$$

ie Q_1^e compr +ve Q_n^e tensile +ve.

or alternatively $\delta I(u) = \delta \left[\frac{1}{2} B(u, u) - L(u) \right] = 0$

$$0 = \delta I(u) = \delta \left[\underbrace{\frac{1}{2} \int_{x_1^e}^{x_n^e} \left(a \left(\frac{du}{dx} \right)^2 + cu^2 \right) dx}_{\text{strain energy due to deformation}} - \underbrace{\int_{x_1^e}^{x_n^e} uq dx - Q_1^e u(x_1^e) - Q_n^e u(x_n^e)}_{\text{work done by loads}} \right]$$

- In deriving weak form / $I(u)$, all BC's for element are "considered/treated/assumed" as NBC's, and thus retained in the weak form. In the next step, while deriving shape/approximating/interpolation f.n.s ϕ_i (aka N_i) we "consider/assume/treat" all BC's as EBC's by considering wffs to be determined, ie u_i^e , as $u_i^e = U^e(x_i^e)$, $u_n^e = U^e(x_n^e)$

- Note that this does not imply that we are applying both NBC's & EBC's which would be fundamentally wrong. In fact in entire step 2 we don't apply any BC's. BC application happens only after assembly, where we identify prescribed displ's (EBCs) & prescribed forces (NBCs), and then solve for unknown displ's, u_j^e , and forces Q_j^e (i.e., reactions).

Substep (ii) Approximating the solution. (deriving approx/shape/interpolation f_{NS})

We assume/consider/treat all BCs as EBCs.

Thus, we require $U(x_i) = u_i^e$, $i = 1, \dots, n$, ($1, \dots, n$ are node numbers of element e). Thus, ϕ_i^e are interpolating f_{NS} .

- ϕ_i^e usually chosen as polynomials, $\therefore$ easy to differentiate and integrate systematically, and interpolation theory using poly's well established.

- Differentiability, completeness, independence, requirements of ϕ_i^e same as for RR (or Weighted Residual) methods, as the case may be.

ϕ_i^e should be interpolant of PV's (u in present case)

Linear $\phi_i \equiv N_i$ $U(x) = a + bx$, $\begin{Bmatrix} u_1^e \\ u_2^e \end{Bmatrix} = \begin{bmatrix} 1 & x_1^e \\ 1 & x_2^e \end{bmatrix} \begin{Bmatrix} a \\ b \end{Bmatrix}$

$$U^e(x) = \frac{x_2^e - x}{x_2^e - x_1^e} u_1^e + \frac{x - x_1^e}{x_2^e - x_1^e} u_2^e = N_1^e(x) u_1^e + N_2^e(x) u_2^e$$

$x \rightarrow$ global coord
 $x_1^e \leq x \leq x_2^e$

$$= \frac{x_{e+1} - x}{h^e} u_1^e + \frac{x - x_e}{h^e} u_2^e$$

$$U^e(\bar{x}) = \frac{h^e - \bar{x}}{h^e} u_1^e + \frac{\bar{x}}{h^e} u_2^e = N_1^e(\bar{x}) u_1^e + N_2^e(\bar{x}) u_2^e$$

$\bar{x} \rightarrow$ local coord. (element)
 $0 \leq \bar{x} \leq h_e$

Quadratic $\phi_i \equiv N_i$: $u(x)^e = a + bx + cx^2$, $\begin{Bmatrix} u_1^e \\ u_2^e \\ u_3^e \end{Bmatrix} = \begin{bmatrix} 1 & x_1^e & (x_1^e)^2 \\ 1 & x_2^e & (x_2^e)^2 \\ 1 & x_3^e & (x_3^e)^2 \end{bmatrix} \begin{Bmatrix} a \\ b \\ c \end{Bmatrix}$

Tedious to solve a, b, c in terms of u_1^e, u_2^e, u_3^e . So use n^{th} order Lagrangian Polynomial (which gives same result), given as

$$N_i(x) = \prod_{j=1, j \neq i}^n \frac{x - x_j}{x_i - x_j}$$

Here onwards superscript 'e' dropped in x_j^e for convenience in writing.

$$N_1(x) = \left(\frac{x - x_2}{x_1 - x_2} \right) \left(\frac{x - x_3}{x_1 - x_3} \right), \quad N_2(x) = \frac{(x - x_1)(x - x_3)}{(x_2 - x_1)(x_2 - x_3)}$$

$$N_3(x) = \frac{(x - x_1)(x - x_2)}{(x_3 - x_1)(x_3 - x_2)}$$

$$x_3 - x_1 = h^e, \quad x_2 - x_1 = \alpha h^e, \quad x_1 - x = \bar{x}$$

when ② at midpoint $\alpha = 1/2$

$$N_1(\bar{x}) = \frac{(x - x_1 - \alpha h^e)(x - x_1 - h^e)}{(\alpha h^e)(-h^e)} = \left(\frac{1 - \bar{x}}{\alpha h^e} \right) \left(\frac{1 - \bar{x}}{h^e} \right) = \left(\frac{1 - \bar{x}}{h^e} \right) \left(\frac{1 - 2\bar{x}}{h^e} \right)$$

$$N_2(\bar{x}) = \frac{(\bar{x})(\bar{x} - h^e)}{(\alpha h^e)(\alpha - 1)h^e} = \frac{1}{\alpha(1 - \alpha)} \frac{\bar{x}}{h^e} \left(\frac{1 - \bar{x}}{h^e} \right) = 4 \left(\frac{\bar{x}}{h^e} \right) \left(\frac{1 - \bar{x}}{h^e} \right)$$

$$N_3(\bar{x}) = \frac{(\bar{x})(\bar{x} - \alpha h^e)}{(h^e)(1 - \alpha)h^e} = -\frac{\alpha}{(1 - \alpha)} \left(\frac{\bar{x}}{h^e} \right) \left(\frac{1 - \bar{x}}{\alpha h^e} \right) = -\left(\frac{\bar{x}}{h^e} \right) \left(\frac{1 - 2\bar{x}}{h^e} \right)$$

Cubic $\phi_i \equiv N_i$ $u_x^e = a + bx + cx^2 + dx^3$, 4x4 system to solve for a, b, c, d . Too tedious!!

so instead, $N_1(x) = \frac{(x - x_2)(x - x_3)(x - x_4)}{(x_1 - x_2)(x_1 - x_3)(x_1 - x_4)}, \quad N_2(x) = \frac{(x - x_1)(x - x_3)(x - x_4)}{(x_2 - x_1)(x_2 - x_3)(x_2 - x_4)}$

$$N_3(x) = \frac{(x - x_1)(x - x_2)(x - x_4)}{(x_3 - x_1)(x_3 - x_2)(x_3 - x_4)}, \quad N_4(x) = \frac{(x - x_1)(x - x_2)(x - x_3)}{(x_4 - x_1)(x_4 - x_2)(x_4 - x_3)}$$

$$x_4 - x_1 = h^e, \quad x_2 - x_1 = \alpha h^e, \quad x_3 - x_1 = \beta h^e, \quad x_1 - x = \bar{x}$$

$$N_1(\bar{x}) = \frac{(x - x_1 - \alpha h^e)(x - x_1 - \beta h^e)(x - x_1 - h^e)}{(-\alpha h^e)(-\beta h^e)(-h^e)} = \left(1 - \frac{\bar{x}}{\alpha h^e}\right) \left(1 - \frac{\bar{x}}{\beta h^e}\right) \left(1 - \frac{\bar{x}}{h^e}\right)$$

$$= \left(1 - \frac{\bar{x}}{h^e}\right) \left(1 - \frac{3\bar{x}}{2h^e}\right) \left(1 - \frac{3\bar{x}}{h^e}\right) \quad \text{for } \alpha = \frac{1}{3}, \beta = \frac{2}{3}$$

$$N_2(\bar{x}) = \frac{(\bar{x})(\bar{x} - \beta h^e)(\bar{x} - h^e)}{(\alpha h^e)(\alpha - \beta)h^e(\alpha - 1)h^e} = -\left(\frac{\bar{x}}{h^e}\right) \left(1 - \frac{\bar{x}}{h^e}\right) \left(\beta - \frac{\bar{x}}{h^e}\right) \frac{1}{\alpha(1-\alpha)(\beta-\alpha)}$$

$$= -9 \left(\frac{\bar{x}}{h^e}\right) \left(1 - \frac{\bar{x}}{h^e}\right) \left(1 - \frac{3\bar{x}}{2h^e}\right) \quad \text{for } \alpha = \frac{1}{3}, \beta = \frac{2}{3}$$

$$N_3(\bar{x}) = \frac{(\bar{x})(\bar{x} - \alpha h^e)(\bar{x} - h^e)}{(\beta h^e)(\beta - \alpha)h^e(\beta - 1)h^e} = -\left(\frac{\bar{x}}{h^e}\right) \left(1 - \frac{\bar{x}}{h^e}\right) \left(\alpha - \frac{\bar{x}}{h^e}\right) \frac{1}{\beta(\beta-\alpha)(1-\beta)}$$

$$= -\frac{9}{2} \left(\frac{\bar{x}}{h^e}\right) \left(1 - \frac{\bar{x}}{h^e}\right) \left(1 - \frac{3\bar{x}}{h^e}\right) \quad \text{for } \alpha = \frac{1}{3}, \beta = \frac{2}{3}$$

$$N_4(x) = \frac{(\bar{x})(\bar{x} - \alpha h^e)(\bar{x} - \beta h^e)}{h^e(1-\alpha)h^e(1-\beta)h^e} = \frac{\bar{x}}{h^e} \left(\alpha - \frac{\bar{x}}{h^e}\right) \left(\beta - \frac{\bar{x}}{h^e}\right) \frac{1}{(1-\alpha)(1-\beta)}$$

$$= \left(\frac{\bar{x}}{h^e}\right) \left(1 - \frac{3\bar{x}}{h^e}\right) \left(1 - \frac{3\bar{x}}{2h^e}\right) \quad \text{for } \alpha = \frac{1}{3}, \beta = \frac{2}{3}$$

$$(i) \quad N_i^e(x_j^e) = N_i(\bar{x}_j) = \delta_{ij}$$

$$(ii) \quad \sum_{i=1}^n N_i^e(x) = \sum_{i=1}^n N_i^e(\bar{x}) = 1 \Rightarrow \sum_{i=1}^n \frac{d}{dx} N_i(x) = \sum_{i=1}^n \frac{d}{d\bar{x}} N_i(\bar{x}) = 0$$

$\rightarrow$ Properties of N_i

(i) $\because N_i$ is an interpolation fn, ie $U(x_i) = U_i^e, i=1, \dots, n$

(ii) $\because U(x) = U(\bar{x})$ must allow for the possibility of a constant solution, $U(x) = U(\bar{x}) = \text{const} = U_1^e = U_2^e = \dots = U_n^e$, which implies property (ii).

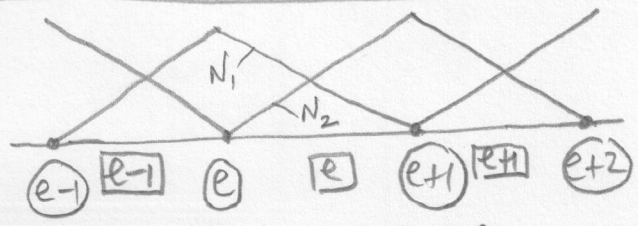
Note: In beams, ^(u → w) the d.o.f at each node are $w, \frac{dw}{dx}$
 so we interpolate PV's w and $\frac{dw}{dx}$ and hence a
 constant w or $\frac{dw}{dx}$ solution won't imply property (ii)

→ • In axial rod, the interpolation of PV's ^(only u) involve only dependent unknown u . These are Lagrange family of interpolation fns and possess C^0 continuity $\because$ only displacements (u) and not its derivatives are used to derive the interpolation fns and only displ and not its derivatives are interpolated by these fns.

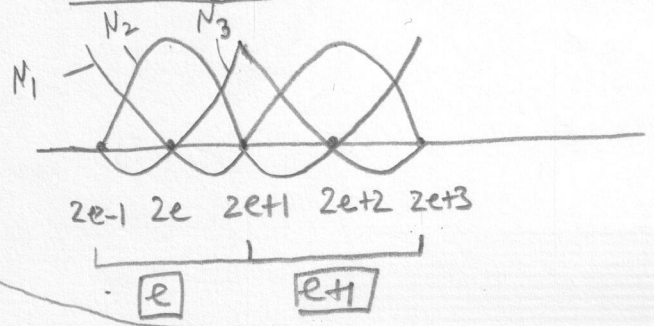
→ • When interpolation fns are derived using dependent variable & its derivatives, they are Hermite family of interpolation fns and possess C^1 continuity. Example beam, where PV's are w, w' , both of which are used to derive interpolation fns and both of which are interpolated by these fns and their 1st derivative.

→ • $C^0 \Rightarrow$ the approximation using interpolation fns yields continuous dependent unknown (eg. u) but not continuous derivatives (eg. u') as in axial rod using Lagrangian
 $C^1 \Rightarrow$ approx yields continuous dep. unknown (eg. w) and its derivatives (eg. w') as in beam using Hermite.

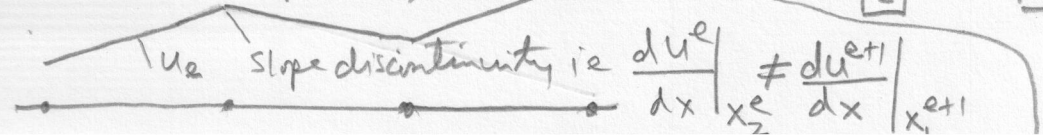
N_i for linear Lagrange element



N_i for quadratic Lagrange element



so for linear interpolation, u^e is



Can derive interpolation functions using property (i)

Quadratic Lagrange

$$N_1 = C \left(\bar{x} - \frac{1}{2} h_e \right) (\bar{x} - h_e); N_1 \Big|_{\bar{x}=0} = 1 \Rightarrow C = \frac{2}{h_e^2}$$

$\because N_1 = 0$ at
 $\bar{x} = \frac{1}{2} h_e$ and h_e

$$\Rightarrow N_1 = \left(1 - \frac{\bar{x}}{h_e} \right) \left(1 - 2 \frac{\bar{x}}{h_e} \right)$$

$$N_2 = C (\bar{x}) (\bar{x} - h_e); N_2 \Big|_{\bar{x} = \frac{h_e}{2}} = 1 \Rightarrow C = -\frac{4}{h_e^2} \Rightarrow N_2 = 4 \left(\frac{\bar{x}}{h_e} \right) \left(1 - \frac{\bar{x}}{h_e} \right)$$

$$N_3 = C (\bar{x}) \left(\bar{x} - \frac{h_e}{2} \right); N_3 \Big|_{\bar{x} = h_e} = 1 \Rightarrow C = \frac{2}{h_e^2} \Rightarrow N_3 = -\left(\frac{\bar{x}}{h_e} \right) \left(1 - 2 \frac{\bar{x}}{h_e} \right)$$

Cubic Lagrange

$$N_1 = C \left(\bar{x} - \frac{h_e}{3} \right) \left(\bar{x} - \frac{2}{3} h_e \right) (\bar{x} - h_e); N_1 \Big|_{\bar{x}=0} = 1 \Rightarrow C = -\frac{9}{2} h_e^3$$

$$\Rightarrow N_1 = \left(1 - \frac{\bar{x}}{h_e} \right) \left(1 - \frac{3}{2} \frac{\bar{x}}{h_e} \right) \left(1 - 3 \frac{\bar{x}}{h_e} \right)$$

$$N_2 = C (\bar{x}) \left(\bar{x} - \frac{2}{3} h_e \right) (\bar{x} - h_e); N_2 \Big|_{\bar{x} = \frac{h_e}{3}} = 1 \Rightarrow C = \frac{27}{2} \frac{1}{h^3}$$

$$\Rightarrow N_2 = 9 \left(\frac{\bar{x}}{h_e} \right) \left(1 - \frac{3}{2} \frac{\bar{x}}{h_e} \right) \left(1 - \frac{\bar{x}}{h_e} \right)$$

$$N_3 = C (\bar{x}) \left(\bar{x} - \frac{1}{3} h_e \right) (\bar{x} - h_e); N_3 \Big|_{\bar{x} = \frac{2}{3} h_e} = 1 \Rightarrow C = -\frac{27}{2} \frac{1}{h^3}$$

$$\Rightarrow N_3 = -9 \left(\frac{\bar{x}}{h_e} \right) \left(1 - \frac{\bar{x}}{h_e} \right) \left(1 - 3 \frac{\bar{x}}{h_e} \right)$$

$$N_4 = C (\bar{x}) \left(\bar{x} - \frac{h_e}{3} \right) \left(\bar{x} - \frac{2}{3} h_e \right); N_4 \Big|_{\bar{x} = h_e} = 1 \Rightarrow C = \frac{9}{2} \frac{1}{h_e^3}$$

$$\Rightarrow N_4 = \left(\frac{\bar{x}}{h_e} \right) \left(1 - \frac{3}{2} \frac{\bar{x}}{h_e} \right) \left(1 - 3 \frac{\bar{x}}{h_e} \right)$$

- Lagrange interpolation fns. readily available in books. They depend only on geometry ^(shape) of element, & nos and location of nodes in element.

Substep (iii) Finite Element Model (relation between PV & SV for element) (17)

Recall $U^e = \sum_{j=1}^n u_j^e \underbrace{N_j^e}_{\equiv \phi_j^e}$. Subst in Weak form of Element

Recall W.F $\rightarrow \int_{x_1^e}^{x_n^e} \left(a \frac{dw}{dx} \frac{du}{dx} + c w u - w q \right) dx - w(x_1^e) Q_1^e - w(x_n^e) Q_n^e$

where, $Q_1^e = -a \frac{du}{dx} \Big|_{x=x_1^e}$, $Q_n^e = a \frac{du}{dx} \Big|_{x=x_n^e}$ $n = \text{nos of nodes in element}$
defined w/ve convention

$$B_{ij}^e \equiv K_{ij}^e = \int_0^{he} \left[a(\bar{x}) \frac{dN_i^e(\bar{x})}{d\bar{x}} \frac{dN_j^e(\bar{x})}{d\bar{x}} + c(\bar{x}) N_i^e(\bar{x}) N_j^e(\bar{x}) \right] d\bar{x}$$

$$F_i^e = \int_0^{he} N_i^e(\bar{x}) q(\bar{x}) d\bar{x} + \sum_{j=1}^n N_i(\bar{x}_j^e) Q_j^e$$

Note: In F_i^e we added applied point sources (ie applied loads) at nodes $2, \dots, (n-1)$, i.e, $Q_2^e, \dots, Q_{n-1}^e$, by representing them as $Q_j^e \delta(\bar{x} - \bar{x}_j)$ in the $q(\bar{x})$ term and using $\int_0^{he} N_i^e(\bar{x}) Q_j^e \delta(\bar{x} - \bar{x}_j) d\bar{x} = N_i^e(\bar{x}_j) Q_j^e$

Also $B_{i0}^e = B(N_i^e, N_0^e)$ not present (ie zero) $\because N_0^e = 0$, ie all BC's considered as NBC's for element.

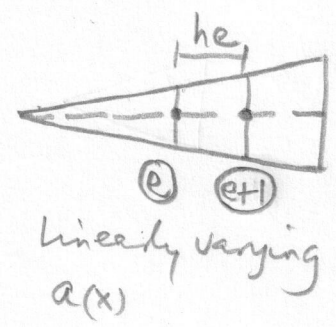
Linear N_i : $a^e(\bar{x}) = a_0$, $c^e(\bar{x}) = c_0$, $q^e(\bar{x}) = q_0$, ie const

$$K^e = \int_0^{he} \left\{ a_0 \begin{bmatrix} -1/he \\ 1/he \end{bmatrix} \begin{bmatrix} -1/he & 1/he \end{bmatrix} + c_0 \begin{bmatrix} 1-\bar{x}/he \\ \bar{x}/he \end{bmatrix} \begin{bmatrix} 1-\bar{x}/he & \bar{x}/he \end{bmatrix} \right\} d\bar{x}$$

$$= \frac{a_0}{he} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} + \frac{c_0 he}{6} \begin{bmatrix} 1-1+\frac{1}{3} & \frac{1}{2}-\frac{1}{3} \\ \frac{1}{2}-\frac{1}{3} & \frac{1}{3} \end{bmatrix} = \frac{c_0 he}{6} \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix}$$

$$F^e = \int_0^{he} \begin{bmatrix} 1 - \bar{x}/he \\ \bar{x}/he \end{bmatrix} q_0 + \begin{bmatrix} \sum_{j=1}^2 (1 - \bar{x}/he) |_{x_j} Q_j \\ \sum_{j=1}^2 (\bar{x}/he) |_{x_j} Q_j \end{bmatrix} = q_0 he \begin{bmatrix} 1 - \frac{1}{2} = \frac{1}{2} \\ \frac{1}{2} \end{bmatrix} + \begin{bmatrix} Q_1 \\ Q_2 \end{bmatrix}$$

If $a(x) = a_0 x$ (tapered section), $a(\bar{x}) = a_0(x_e + \bar{x})$
 then first term in K^e becomes,



$$\int_0^{he} \frac{a_0(x_e + \bar{x})}{he^2} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} d\bar{x} = \frac{a_0}{he} \left(x_e + \frac{he}{2} \right) \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

average of areas at x_e, x_{e+1} .

$$K^e \begin{Bmatrix} u_1^e \\ u_2^e \end{Bmatrix} = F^e$$

→ linear element Force-displ (SV-PV) relation

Quadratic Ni : $a(x) = a_0, c(x) = c_0, q(x) = q_0$

$$K^e = \int_0^{he} \left\{ \begin{array}{l} \frac{a_0}{he^2} \begin{bmatrix} -1 + \frac{2\bar{x}}{he} & -2 + \frac{2\bar{x}}{he} \\ 4 - \frac{4\bar{x}}{he} & -4\bar{x}/he \\ -1 + \frac{2\bar{x}}{he} & \frac{2\bar{x}}{he} \end{bmatrix} \begin{bmatrix} -3 + \frac{4\bar{x}}{he} & 4 - \frac{8\bar{x}}{he} & -1 + \frac{4\bar{x}}{he} \end{bmatrix} \\ c_0 \begin{bmatrix} 1 - \frac{3\bar{x}}{he} + 2\left(\frac{\bar{x}}{he}\right)^2 \\ \frac{4\bar{x}}{he} - 4\left(\frac{\bar{x}}{he}\right)^2 \\ -\frac{\bar{x}}{he} + 2\left(\frac{\bar{x}}{he}\right)^2 \end{bmatrix} \begin{bmatrix} \text{transpose} \end{bmatrix} \end{array} \right\} d\bar{x}$$

$$-\frac{1}{2} + \frac{5}{3} - \frac{8}{4} + \frac{4}{5} = -\frac{1}{30}$$

$$= \frac{a_0}{he} \begin{bmatrix} \frac{7}{3} & -\frac{8}{3} & \frac{1}{3} \\ \frac{16}{3} & -\frac{8}{3} & \frac{7}{3} \\ \frac{7}{3} & -\frac{8}{3} & \frac{1}{3} \end{bmatrix} + c_0 he \begin{bmatrix} \frac{2}{15} & \frac{1}{15} \\ \frac{8}{15} & \frac{1}{15} \end{bmatrix}$$

Symm

$$F^e = \int_0^{he} \begin{bmatrix} 1 - 3\frac{\bar{x}}{he} + 2\left(\frac{\bar{x}}{he}\right)^2 \\ 4\frac{\bar{x}}{he} - 4\left(\frac{\bar{x}}{he}\right)^2 \\ -\frac{\bar{x}}{he} + 2\left(\frac{\bar{x}}{he}\right)^2 \end{bmatrix} q_0 d\bar{x} + \begin{Bmatrix} Q_1^e \\ Q_2^e \\ Q_3^e \end{Bmatrix} = \begin{Bmatrix} 1 - \frac{3}{2} + \frac{2}{3} = \frac{1}{6} \\ \frac{4}{2} - \frac{4}{3} = \frac{2}{3} \\ -\frac{1}{2} + \frac{2}{3} = \frac{1}{6} \end{Bmatrix} q_0 he + \begin{Bmatrix} Q_1^e \\ Q_2^e \\ Q_3^e \end{Bmatrix}$$

$$K^e \begin{Bmatrix} u_1^e \\ u_2^e \\ u_3^e \end{Bmatrix} = F^e$$

Quadratic element Force-Displ (SV-PV) relation

Note: if you distribute q_0 as per linear element of length $he/2$ you get

Note: $\sum_{i=1}^n f_i^e = \left(\frac{1}{6} + \frac{2}{3} + \frac{1}{6}\right) q_0 he = q_0 he$
 also evident from $f_i = \int N_i q dx$
 and $\sum_{i=1}^n N_i^e = 1$. This means that source when distributed to element nodes should add up to source integrated over element.

$\begin{Bmatrix} \frac{1}{2} \cdot \frac{1}{2} \\ \frac{1}{2} \cdot \frac{1}{2} \times 2 \\ \frac{1}{2} \cdot \frac{1}{2} \end{Bmatrix} q_0 he$ which is wrong!!

Step 3 Assembly (Structure SV-PV relation)

Continuity of PV's: eg, $u_n^e = u_1^{e+1} = u_N$ for sequentially arranged axial elements.

Balance of SV's: eg. $Q_n^e + Q_1^{e+1} = Q_N$
 $N = \text{global number of } n^{\text{th}} \text{ (last) node of } e^{\text{th}} \text{ element}$
 $= ne - (e-1) = (n-1)e + 1$

don't confuse with N-term approximation in traditional variational methods

$Q_N = \text{applied point source (load) at global node } N$.

Actually balance of SV is equivalent to continuity of SV.

i.e., $\left(\frac{adu}{dx}\right)_n^e = \left(\frac{adu}{dx}\right)_1^{e+1} \Rightarrow Q_n^e = -Q_1^{e+1}, Q_n^e + Q_1^{e+1} = 0$

if $Q_N = 0$, i.e. no point source applied

Note: $\left(\frac{adu}{dx}\right)_n^e \neq \left(\frac{adu}{dx}\right)_1^{e+1}$, i.e., gradient of approx soln is not continuous across element end nodes

Adding n^{th} equil eqn of e^{th} element and 1^{st} equil eqn of $(e+1)^{th}$ element (assuming sequential global node numbering and both elements n -noded)

$$K_{n1}^e u_1^e + \dots + K_{nn}^e u_n^e + K_{11}^{e+1} u_1^{e+1} + \dots + K_{in}^{e+1} u_i^{e+1} = f_n^e + f_1^{e+1} + Q_n^e + Q_1^{e+1}$$

and introducing interelement continuity of PV's & equilibrium of SV's,

$$0 \dots 0 \quad K_{n1}^e \quad K_{n2}^e \dots K_{nj}^e \dots (K_{nn}^e + K_{11}^{e+1}) \quad K_{12}^{e+1} \dots K_{ij}^{e+1} \dots K_{in}^{e+1} \quad 0 \dots 0$$

u_1
 $\vdots$
 $u_{N-(n-1)}$
 u_{N-n+2}
 $\vdots$
 u_N
 u_{N+1}
 $\vdots$
 $u_{N+(n-1)}$
 $\vdots$
 $u_{E(n-1)+1}$

are applied point sources or reactions at element end nodes
 applied point sources at element intermediate nodes

equivalent nodal loads due to $q(x)$.

=

$q(x)$ can also include applied point loads at nodes, i.e. $\delta(x-x^*)$ in which case the corresponding Q 's should be made zero.

$f_1^1 \rightarrow f_1$

$\vdots$

$f_j^1 \rightarrow f_j$

$\vdots$

$f_n^1 + f_1^2 \rightarrow f_{(n-1)e+1}^1$

$\vdots$

$f_j^2 \rightarrow f_{(n-1)(e-1)+1+j-1}^2$

$\vdots$

$f_n^2 + f_1^3 \rightarrow f_{(n-1)e+1}^2$

$\vdots$

$f_n^{e-1} + f_1^e \rightarrow f_{(n-1)(e-1)+1}^e = N-(n-1)$

$f_2^e \rightarrow f_{N-n+2}$

$\vdots$

$f_n^e + f_1^{e+1} \rightarrow f_{(n-1)e+1}^e = N$

$f_2^{e+1} \rightarrow f_{N+1}$

$\vdots$

$[Q_1^1]$

$\vdots$

Q_j^1

$\vdots$

$Q_n^1 + Q_1^2 \rightarrow Q_{(n-1)e+1}^1$

$\vdots$

$Q_j^2 \rightarrow Q_{(n-1)(e-1)+j}^2$

$\vdots$

$Q_n^2 + Q_1^3 \rightarrow Q_{(n-1)e+1}^2$

$\vdots$

$Q_n^{e-1} + Q_1^e \rightarrow Q_{N-(n-1)}$

$Q_2^e \rightarrow Q_{N-n+2}$

$\vdots$

$Q_n^e + Q_1^{e+1} \rightarrow Q_N$

$Q_2^{e+1} \rightarrow Q_{N+1}$

$\vdots$

So for mesh with 'E' linear (2-noded) elements,

(21)

$$\begin{bmatrix}
 K_{11}^1 & K_{12}^1 \\
 K_{21}^1 + (K_{22}^1 + K_{11}^2) & K_{12}^2 \\
 K_{21}^2 & (K_{22}^2 + K_{11}^3) & K_{12}^3 \\
 \vdots & \vdots & \vdots \\
 K_{21}^{e-1} & (K_{22}^{e-1} + K_{11}^e) & K_{12}^e \\
 K_{21}^e & (K_{22}^e + K_{11}^{e+1}) & K_{12}^{e+1} \\
 \vdots & \vdots & \vdots \\
 K_{21}^{E-1} & (K_{22}^{E-1} + K_{11}^E) & K_{12}^E \\
 K_{21}^E & K_{22}^E
 \end{bmatrix}$$

equivalent
nodal loads
due to $q(x)$
(which can
be due to
point sources
 $q(x) = \delta(x-x^*)$)

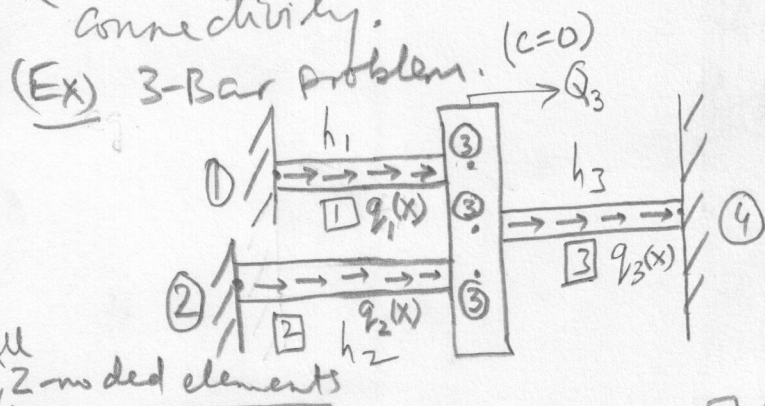
$$= \begin{bmatrix}
 f_1^1 \rightarrow f_1 \\
 f_2^1 + f_1^2 \\
 f_2^2 + f_1^3 \\
 \vdots \\
 f_2^{e-1} + f_1^e \\
 f_2^e + f_1^{e+1} \\
 \vdots \\
 f_2^{E-1} + f_1^E \\
 f_2^E
 \end{bmatrix}$$

$f_E \leftarrow f_2^{E-1} + f_1^E$
 $f_{E+1} \leftarrow f_2^E$

$$+ \begin{bmatrix}
 Q_1^1 \rightarrow Q_1 \\
 Q_2^1 + Q_1^2 \\
 Q_2^2 + Q_1^3 \\
 \vdots \\
 Q_2^{e-1} + Q_1^e \\
 Q_2^e + Q_1^{e+1} \\
 \vdots \\
 Q_2^{E-1} + Q_1^E \rightarrow Q_E \\
 Q_2^E \rightarrow Q_{E+1}
 \end{bmatrix}$$

applied point sources
or reactions at element
end nodes,
i.e., $1, \dots, (k-1)+1$
 $\downarrow \quad \downarrow$
 $E \quad 2$
i.e., $1, \dots, E+1$

In general (ie when elements are not sequentially arranged), assembly follows same method as for direct stiffness method, ie as per d.o.f numbering and connectivity.



connectivity matrix

$$G = \begin{matrix} & \text{node} & 1 & 2 \\ \text{element} & & & \\ \boxed{1} & & 1 & 3 \\ \boxed{2} & & 2 & 3 \\ \boxed{3} & & 3 & 4 \end{matrix} \rightarrow \text{connectivity matrix}$$

G_{ij} = global node number of j^{th} node of i^{th} element.

all 2-noded elements

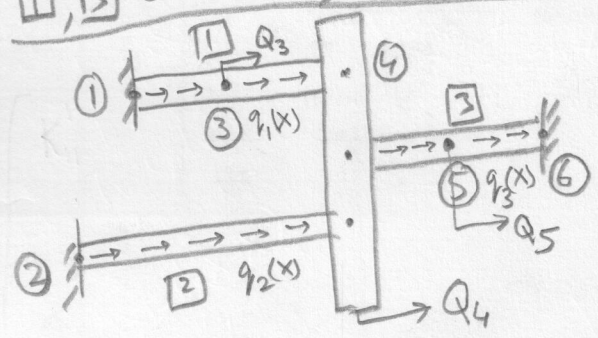
$$\begin{bmatrix} K_{11}^1 & 0 & K_{12}^1 \\ & K_{11}^2 & K_{12}^2 \\ & & K_{22}^1 + K_{22}^2 + K_{11}^3 \end{bmatrix}$$

symm

$\boxed{1}, \boxed{3}$ 3-noded, $\boxed{2}$ 2-noded:

$$\begin{bmatrix} 0 \\ 0 \\ K_{12}^3 \\ K_{22}^3 \end{bmatrix} \begin{cases} u_1^1 \equiv u_1 \\ u_1^2 \equiv u_2 \\ u_2^1 = u_2^2 = u_1^3 \equiv u_3 \\ u_2^3 \equiv u_4 \end{cases} = \begin{cases} f_1^1 \equiv f_1 \\ f_1^2 \equiv f_2 \\ f_2^1 + f_2^2 + f_1^3 \equiv f_3 \\ f_2^3 \equiv f_4 \end{cases} + \begin{cases} Q_1^1 \equiv Q_1 \\ Q_1^2 \equiv Q_2 \\ Q_2^1 + Q_2^2 + Q_1^3 \equiv Q_3 \\ Q_2^3 \equiv Q_4 \end{cases}$$

(*)



$K = B$

$$G = \begin{matrix} & 1 & 2 & 3 \\ \boxed{1} & 1 & 3 & 4 \\ \boxed{2} & 2 & - & 4 \\ \boxed{3} & 4 & 5 & 6 \end{matrix}$$

$$\begin{bmatrix} K_{11}^1 & 0 & K_{12}^1 & K_{13}^1 \\ & K_{11}^2 & 0 & K_{12}^2 \\ & & K_{22}^1 & K_{23}^1 \\ & & & K_{33}^1 + K_{22}^2 + K_{11}^3 \end{bmatrix}$$

symm

$$\begin{bmatrix} 0 \\ 0 \\ 0 \\ K_{12}^3 \\ K_{23}^3 \\ K_{33}^3 \end{bmatrix} \begin{cases} u_1^1 \equiv u_1 \\ u_1^2 \equiv u_2 \\ u_2^1 \equiv u_3 \\ u_3^1 = u_2^2 = u_1^3 \equiv u_4 \\ u_2^3 \equiv u_5 \\ u_3^3 \equiv u_6 \end{cases} = \begin{cases} f_1^1 \equiv f_1 \\ f_1^2 \equiv f_2 \\ f_2^1 \equiv f_3 \\ f_3^1 + f_2^2 + f_1^3 \equiv f_4 \\ f_2^3 \equiv f_5 \\ f_3^3 \equiv f_6 \end{cases}$$

$$+ \begin{cases} Q_1^1 \equiv Q_1 \\ Q_1^2 \equiv Q_2 \\ Q_2^1 \equiv Q_3 \\ Q_3^1 + Q_2^2 + Q_1^3 \equiv Q_4 \\ Q_2^3 \equiv Q_5 \\ Q_3^3 \equiv Q_6 \end{cases}$$

here Q_3, Q_5 (circled) are always known applied point sources.

Applying BC's

For the 3-bar problem with loads as shown,

All 2-noded elements: $u_1 = u_2 = u_4 = 0$, $Q_3 = \text{given}$; unknowns are u_3, Q_1, Q_2, Q_4

[1], [3] 3-noded, [2] 2-noded: $u_1 = u_2 = u_6 = 0$, $Q_3, Q_4, Q_5 = \text{given}$; unknowns are $u_3, u_4, u_5, Q_1, Q_2, Q_6$.

$$\begin{aligned} \begin{Bmatrix} f_1^e \\ f_2^e \end{Bmatrix} &= \int_0^{h_e} \begin{Bmatrix} N_1^e(\bar{x}) \\ N_2^e(\bar{x}) \end{Bmatrix} q_e(\bar{x}) d\bar{x} \quad \text{for } e=1, 2, 3 \\ \begin{Bmatrix} f_1^e \\ f_2^e \\ f_3^e \end{Bmatrix} &= \int_0^{h_e} \begin{Bmatrix} N_1^e(\bar{x}) \\ N_2^e(\bar{x}) \\ N_3^e(\bar{x}) \end{Bmatrix} q_e(\bar{x}) d\bar{x} \quad \text{For } e=2. \end{aligned}$$

Solution

$$(*) \leftarrow \begin{bmatrix} K_{II} & K_{II} \\ K_{II} & K_{II} \end{bmatrix} \begin{Bmatrix} U_I \\ U_{II} \end{Bmatrix} = \begin{Bmatrix} f \\ f \end{Bmatrix} + \begin{Bmatrix} Q_I \\ Q_{II} \end{Bmatrix}, \quad U_I, Q_{II} \text{ are unknown PV's \& SV's respectively.}$$

i.e., just as in Stiffness Method.

First compute U_I , then Q_{II} , using Equil. eqns (*). Alternatively

$\frac{du}{dx} = \frac{dU^e}{dx} = \sum_{j=1}^n u_j^e \frac{dN_j^e}{dx}$ can be used to compute Q_{II} , i.e.,

$$Q_1 = Q_1' = - \left[a_1(\bar{x}) \sum_{j=1}^n u_j^1 \frac{dN_j^1}{d\bar{x}} \right]_{\bar{x}=0}, \quad Q_2 = Q_2' = - \left[a_2(\bar{x}) \sum_{j=1}^n u_j^2 \frac{dN_j^2}{d\bar{x}} \right]_{\bar{x}=0}$$

(element no)

for 2-noded [3] $ie n=2 \rightarrow Q_4 = Q_2^3 = \left[a_3(\bar{x}) \sum_{j=1}^n u_j^3 \frac{dN_j^3}{d\bar{x}} \right]_{\bar{x}=h_3}$

or $Q_6 = Q_3^3$ (for 3-noded [3]) $ie n=3$

The above way of computing SV's by definition, i.e. $Q = \pm \frac{adu}{dx}$ is less accurate than computing thru equilibrium eqn (*), but it is used in FEM due to ease of computation. Note

that $\left. \frac{dU^e}{dx} \right|_{x=x_n^e} \neq \left. \frac{dU^{e+1}}{dx} \right|_{x=x_{n+1}^{e+1}} \therefore \text{Lagrange}$ (29)

elements are C^0 continuous, i.e. they interpolate dependent variable but not its derivatives.

This inaccuracy of computing SVs by definition decreases as nos of elements 'E' and/or degree of interpolation 'n' increase.

For (EX) problem, if $q_1 = q_2 = q_3 = 0$ (no distributed ^{considering} all-2-noded elements we have, ^{from definition} sources/sinks),

$$Q_1 = Q_1^1 = - \left(EA \cdot \frac{U_3 - U_1}{h_1} \right)_{\bar{x}=0}^0 = - \frac{EA}{h_1} U_3 = K_{12}^1 U_3$$

$$Q_2 = Q_1^2 = - \left(EA \cdot \frac{U_3 - U_2}{h_2} \right)_{\bar{x}=0}^0 = - \frac{EA}{h_2} U_3 = K_{12}^2 U_3$$

$$Q_3 = Q_2^3 = \left(EA \cdot \frac{U_4 - U_3}{h_3} \right)_{\bar{x}=0}^0 = - \frac{EA}{h_3} U_3 = K_{12}^3 U_3$$

which is exactly same as from equilibrium (★ p.22)

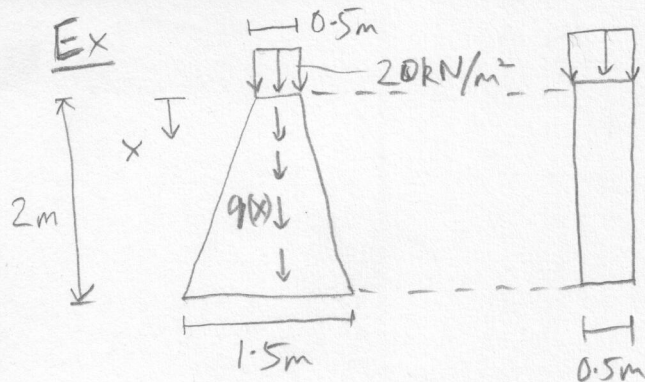
However, this match of SV obtained via definition and via equilibrium, is not to be expected, with latter being more accurate. In fact if q_1, q_2, q_3 were non-zero, this match would not have occurred.

- can repeat these steps using Galerkin or any weighted residual method
- Interpolation $f \approx$ depend on element geometry, number & position of nodes.
- Same formulation holds for class of problems (ie governed by $-(au')' + cu - q = 0$, in present case)
- Integrations for K, f , may need to be done numerically if they are complicated.
- Point sources at intermediate nodes, can be handled thru q terms or $q(\bar{x}) = P\delta(\bar{x} - \bar{x}_p)$ terms, but not both. For point sources not at nodes we handle thru q , ie $\int_0^{he} [N_1 \ N_2 \dots N_n]^T P \delta(\bar{x} - \bar{x}_p) d\bar{x}$

$$= P [N_1, \dots, N_n]^T \Big|_{\bar{x} = \bar{x}_p}$$

Thus for linear element, due to point source at $\bar{x}_p = \alpha h$ we have $f = [1 - \alpha, \alpha]^T P$

- Errors arise due to domain approximation, computation of K_{ij}, f_i (numerical integration), and solution (numerical soln of simultaneous equations).
- K is sparse due to FE model where non-zero K_{ij} occurs only at entries corresponding to an element d.o.f. Further in given problem K is symmetric. Both these result in efficient numerical solution of eqn'd. eqns.



Concrete Pier of bridge with deck load (DL+LL).

Concrete weighs $25 \text{ kN/m}^3 = \rho$

$$E = 28 \times 10^6 \text{ N/m}^2$$

Model as 1-D problem

$$Q_0 = 20 \cdot (0.5)^2 = 5 \text{ kN}$$

$$q(x) = \underbrace{0.5(1+x)0.5}_{A(x)} \cdot \underbrace{25}_{\rho} = 6.25(1+x) \text{ kN/m}$$

$$\frac{d}{dx} \left(EA \frac{du}{dx} \right) + q(x) = 0, \quad EA = E \cdot 0.25(1+x)$$

Use linear elements.

$$\text{WF: } - \int_0^{h_e} EA \frac{du}{dx} \frac{du}{dx} dx + \int_0^{h_e} w q dx + \underbrace{W(x_2)}_{N_i} Q_2^e + \underbrace{W(x_1)}_{N_i} Q_1^e = 0, \quad i=1,2$$

$$[B_{ij}]^e = [K_{ij}]^e = \int_0^{h_e} \frac{0.25E(1+x)}{h_e^2} \begin{bmatrix} -1/h_e \\ 1/h_e \end{bmatrix} \begin{bmatrix} -1/h_e & 1/h_e \end{bmatrix} d\bar{x}$$

$$= 0.25E \left[(1+x_e)h_e + \frac{h_e^2}{2} \right] \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} = 0.25(x_{e+1} - x_e) \left(1 + \frac{x_{e+1} + x_e}{2} \right) \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

$$\{F^e\} = \int_0^{h_e} \begin{bmatrix} 1 - \frac{\bar{x}}{h_e} \\ \frac{\bar{x}}{h_e} \end{bmatrix} 6.25(1+x_{e+1} + \bar{x}) d\bar{x} + \begin{bmatrix} Q_1^e \\ Q_2^e \end{bmatrix}$$

$$= \left\{ (1+x_e)h_e + \frac{h_e^2}{2} - (1+x_e) \frac{h_e}{2} - \frac{h_e^2}{3} \right\} + \begin{bmatrix} Q_1^e \\ Q_2^e \end{bmatrix}$$

$$6.25 \left\{ (1+x_e) \frac{h_e}{2} + \frac{h_e^2}{3} \right\}$$

For 2 elements, $x_1^1 = 0, x_2^1 = 1 = x_1^2, x_2^2 = 2,$

$$0.25E \begin{bmatrix} 3/2 & -3/2 & 0 & 0 \\ -3/2 & 3/2 + 5/2 & -5/2 & 0 \\ 0 & -5/2 & 5/2 & 0 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \\ u_4 \end{Bmatrix} = 6.25 \begin{bmatrix} 2/3 \\ 5/6 + 7/6 \\ 4/3 \end{bmatrix} + \begin{bmatrix} Q_1^1 \\ Q_2^1 + Q_2^2 \\ Q_2^2 \end{bmatrix}$$

$\downarrow$
 $= 0$

(27)

$$0.25E \begin{bmatrix} 1.5 & -1.5 \\ -1.5 & 4 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} = \frac{6.25}{6} \begin{Bmatrix} 4 \\ 12 \end{Bmatrix} + \begin{Bmatrix} 5 \\ 0 \end{Bmatrix}$$

$$u_1 = 2.111 \times 10^{-6} \text{ m}, \quad u_2 = 1.238 \times 10^{-6} \text{ m}, \quad Q_2^2 = Q_3 = -E \cdot \frac{5}{2} u_2 - 6.25 \cdot \frac{4}{3}$$

$$= -29.998$$

$$\text{ie } \approx 30 \text{ kN (C).}$$

$$\text{Exact solution: } u(x) = \frac{1}{E} \left[56.25 - 6.25(1+x)^2 - 7.5 \ln \left(\frac{1+x}{3} \right) \right]$$

$$u(0) = 2.08 \times 10^{-6} \text{ m}, \quad u(1) = 1.225 \times 10^{-6} \text{ m}.$$