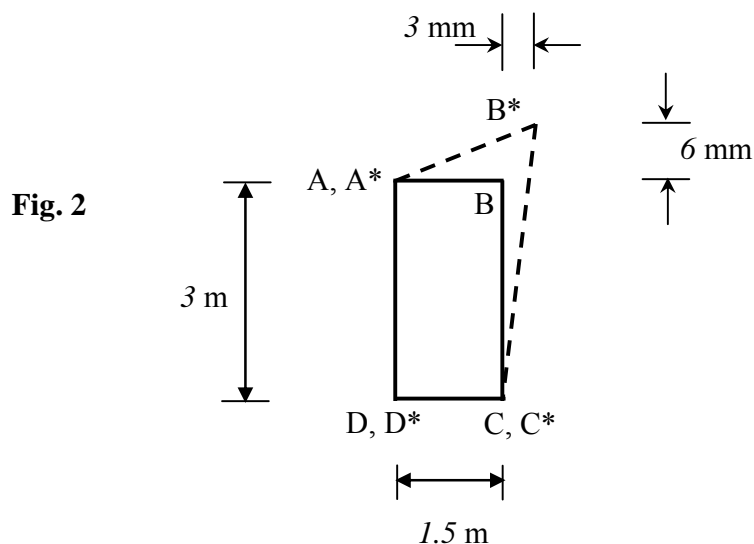


- Equal marks for each question
- Show all working.
- Attempt all parts of a question in a contiguous manner, i.e., don't scatter parts of the same question all over the answerbook.
- **Only one attempt per question will be graded.** So cancel out any attempt you do not want graded. The first not-cancelled attempt will be graded by default.

1. The principal stresses at a point P are 4, 5, 6. **Determine the unit normal for the plane(s) on which the normal stress is 5 and shear stress is $\frac{1}{2}$.**
2. The rectangular plate shown in **Fig. 2**, is loaded so that it is in a state of plane strain (i.e., $\varepsilon_{13} = \varepsilon_{23} = \varepsilon_{33} = 0$). The deformed plate is shown by dotted straight lines. Assume small displacement gradients. **At corner B determine:**
 - a) **Maximum and minimum engineering extensional strains and the direction of the corresponding line elements on which they act.**
 - b) **Maximum change in angle between two elements originally at right angles. Also sketch the direction of these two elements.**



3. At a point on the free surface of an alloy steel ($E=200000$ MPa, $\nu = 0.3$), a strain rosette measures infinitesimal extensional strains of $1000E-6$, $2000E-6$, and $1200E-6$ at respective angles of 0° , 60° and 120° with respect to x axis (measured counterclockwise). Design considerations limit the maximum normal stress to 510 MPa and maximum shearing stress to 275 MPa, the maximum normal strain to $2200E-6$ and maximum shearing strain to $2500E-6$. **What is your evaluation of the design?**

4. A body is subject to uniform pressure such that the state of stress throughout the body is $\sigma_{xx} = \sigma_{yy} = \sigma_{zz} = -p$, $\sigma_{xy} = \sigma_{yz} = \sigma_{zx} = 0$. Assume the elastic constants as E and ν . Given that the displacement and the rotation at the origin are zero, **determine the displacements as a function of x, y, z .**

P1: Refer everything to p-coord system. So,


$$\underline{\underline{\sigma}} = \begin{pmatrix} 4 & 0 & 0 \\ 0 & 5 & 0 \\ 0 & 0 & 6 \end{pmatrix}$$

$$N = \underline{n}^T \underline{\underline{\sigma}} \underline{n} = 4n_1^2 + 5n_2^2 + 6n_3^2 = 5 \rightarrow \textcircled{1}$$

$$S^2 = \sigma^2 - N^2 = (\sigma_{11} n_1)^2 + (\sigma_{22} n_2)^2 + (\sigma_{33} n_3)^2 - N^2$$

$$\Rightarrow \left(\frac{1}{2}\right)^2 = 4^2 n_1^2 + 5^2 n_2^2 + 6^2 n_3^2 - 5^2 \rightarrow \textcircled{2}$$

$$n_1^2 + n_2^2 + n_3^2 = 1 \rightarrow \textcircled{3}$$

Sol. of ①-③ is $n_1 = \pm \frac{1}{2\sqrt{2}}$, $n_2 = \pm \frac{\sqrt{3}}{2}$, $n_3 = \pm \frac{1}{2\sqrt{2}}$ 

i.e., 8 planes exist (only 4 of them are distinct).

P4

Note: stresses, hence strains are uniform (constant, ^{spatially})

$$e_{xx} = \frac{1+\nu}{E} \sigma_{xx} - \frac{\nu}{E} (\sigma_{xx} + \sigma_{yy} + \sigma_{zz}) = \left(\frac{1-2\nu}{E}\right)(-p)$$

$$= e_{yy} = e_{zz}$$

$$e_{xx} = u_{x,x} \Rightarrow u_x = \frac{(2\nu-1)p}{E} x + f(y,z)$$

$$e_{yy} = u_{y,y} \Rightarrow u_y = \frac{(2\nu-1)p}{E} y + g(x,z)$$

$$e_{zz} = u_{z,z} \Rightarrow u_z = \frac{(2\nu-1)p}{E} z + h(x,y)$$

$$e_{xy} = \frac{1}{2}(u_{x,y} + u_{y,x}) = 0 = \frac{1}{2}(f_{,y} + g_{,x}) \rightarrow \textcircled{1}$$

$$e_{yz} = \frac{1}{2}(u_{y,z} + u_{z,y}) = 0 = \frac{1}{2}(g_{,z} + h_{,y}) \rightarrow \textcircled{2}$$

$$e_{zx} = \frac{1}{2}(u_{z,x} + u_{x,z}) = 0 = \frac{1}{2}(h_{,x} + f_{,z}) \rightarrow \textcircled{3}$$

$$\frac{\partial \textcircled{1}}{\partial z} + \frac{\partial \textcircled{2}}{\partial x} = 0 \text{ gives } f_{,zy} + g_{,xz} + g_{,zx} + h_{,xy} = 0 \rightarrow \textcircled{4}$$

Insert $\textcircled{3}$ in $\textcircled{4}$, get $(f_{,z} + h_{,x})_{,y} + 2g_{,xz} = 0$
 $= 0$ from $\textcircled{3}$

$$\Rightarrow g_{,xz} = 0 \Rightarrow g = g_1(x) + g_2(z) \rightarrow \textcircled{5}$$

Similarly you can obtain, $f_{,yz} = 0 \Rightarrow f = f_1(y) + f_2(z) \rightarrow \textcircled{6}$

& $h_{,xy} = 0 \Rightarrow h = h_1(x) + h_2(y) \rightarrow \textcircled{7}$

$\textcircled{5}, \textcircled{6}$ in $\textcircled{1}$ gives $\rightarrow f_1' + g_1' = 0 \Rightarrow f_1' = -g_1' = c_1$ (const)
 $\Rightarrow f_1 = c_1 y + k_1, g_1 = -c_1 x + k_2$ (k_1, k_2 are const).

$\textcircled{5}, \textcircled{7}$ in $\textcircled{2}$ gives $\rightarrow g_2' + h_2' = 0 \Rightarrow g_2' = -h_2' = c_2$
 $\Rightarrow g_2 = c_2 z + k_3, h_2 = -c_2 y + k_4$

$\textcircled{6}, \textcircled{7}$ in $\textcircled{3}$ gives $\rightarrow f_2' + h_1' = 0 \Rightarrow f_2' = -h_1' = c_3$
 $\Rightarrow f_2 = c_3 z + k_5, h_1 = -c_3 x + k_6$

So, $u_x = cx + c_1 y + c_3 z + (k_1 + k_5)$

$u_y = cy + c_2 z - c_1 x + (k_3 + k_2)$

$u_z = cz - c_3 x - c_2 y + (k_6 + k_4)$

From zero-displ at origin $\rightarrow (k_1 + k_5) = (k_3 + k_2) = (k_6 + k_4) = 0$

From zero rotation at origin,

$2\omega_{xy} = u_{x,y} - u_{y,x} = 2c_1 = 0$ at origin $\Rightarrow c_1 = 0 \Rightarrow \omega_{xy} = 0$ everywhere

$2\omega_{yz} = u_{y,z} - u_{z,y} = 2c_2 = 0$ at origin $\Rightarrow c_2 = 0 \Rightarrow \omega_{yz} = 0$ everywhere

$2\omega_{zx} = u_{z,x} - u_{x,z} = -2c_3 = 0$ at origin $\Rightarrow c_3 = 0 \Rightarrow \omega_{zx} = 0$ everywhere

\Rightarrow rotation is zero everywhere.

$u_x = cx \quad u_y = cy \quad u_z = cz \quad c = \frac{(2\nu - 1)}{2} \Delta$

P2

$$\begin{aligned}
 u_1 &= a_1 x_1 + b_1 x_2 + c_1 x_1 x_2 \\
 u_2 &= a_2 x_1 + b_2 x_2 + c_2 x_1 x_2
 \end{aligned}
 \left. \vphantom{\begin{aligned} u_1 \\ u_2 \end{aligned}} \right\} \text{can write directly from Tutorial \#2.}$$

Take origin at D.

$$\begin{aligned}
 \text{At C: } u_1 = 0 = 1.5a_1 & \qquad \text{At A: } u_1 = 0 = 3b_1 \\
 u_2 = 0 = 1.5a_2 & \qquad u_2 = 0 = 3b_2
 \end{aligned}$$

$$\begin{aligned}
 \text{At B: } u_1 = 3E-3 = (1.5)(3)c_1 & \Rightarrow c_1 = \frac{2}{3}E-3 / m \\
 u_2 = 6E-3 = (1.5)(3)c_2 & \Rightarrow c_2 = \frac{4}{3}E-3 / m
 \end{aligned}$$

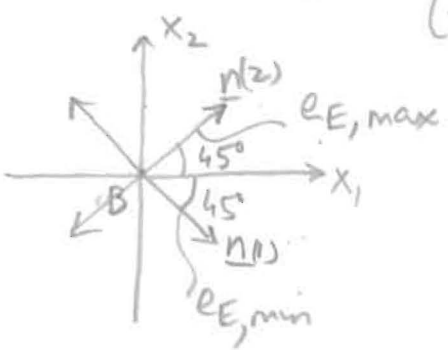
$$\begin{aligned}
 \Rightarrow u_1 &= \left(\frac{2}{3}E-3\right)x_1 x_2 & e_{11} &= u_{1,1} = c_1 x_2 = 2E-3 \\
 u_2 &= \left(\frac{4}{3}E-3\right)x_1 x_2 & e_{22} &= u_{2,2} = c_2 x_1 = 2E-3 \\
 & & e_{12} &= \frac{1}{2}(u_{1,2} + u_{2,1}) = \frac{1}{2}(c_1 x_1 + c_2 x_2) \\
 & & &= \frac{1}{2}\left(\frac{2}{3} \cdot 1.5 + \frac{4}{3} \cdot 3\right) \cdot E-3 = 2.5E-3
 \end{aligned}$$

$$\text{At B, } \underline{e} = \begin{pmatrix} 2 & 2.5 \\ 2.5 & 2 \end{pmatrix} \times 10^{-3}$$

$$\Rightarrow (2-\lambda)^2 - 2.5^2 = (-0.5-\lambda)(4.5-\lambda) = 0$$

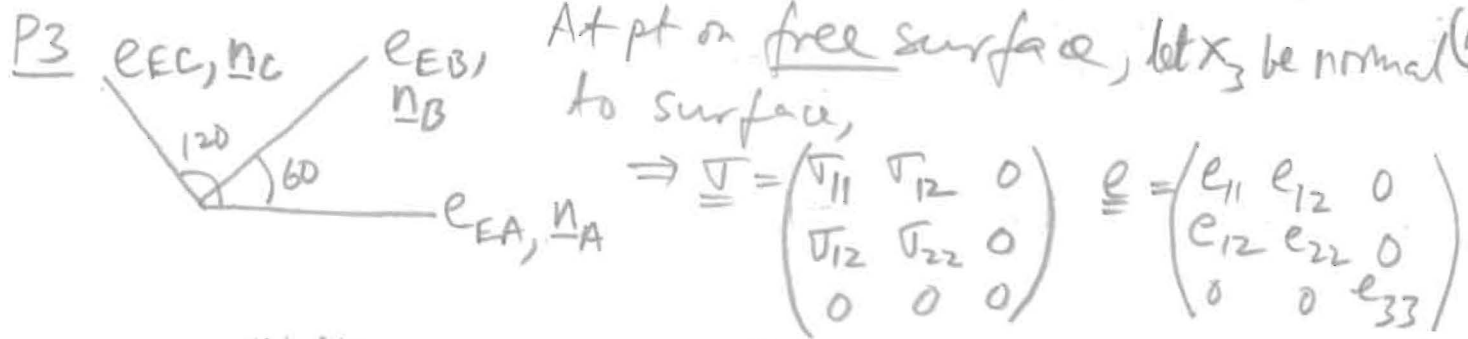
$$\lambda(1) = -0.5E-3 = e_{E, \min} \quad ; \quad \lambda(2) = 4.5E-3 = e_{E, \max}$$

$$\underline{n}(1) = \begin{Bmatrix} 1 \\ -1 \end{Bmatrix} \equiv \begin{Bmatrix} 1/\sqrt{2} \\ -1/\sqrt{2} \end{Bmatrix} \quad ; \quad \underline{n}(2) = \begin{Bmatrix} 1 \\ 1 \end{Bmatrix} \equiv \begin{Bmatrix} 1/\sqrt{2} \\ 1/\sqrt{2} \end{Bmatrix}$$



$$e_{12, \max} = \left| \frac{\lambda(1) - \lambda(2)}{2} \right| = 2.5E-3 \approx \frac{\cos \theta}{2}$$

$$\begin{aligned}
 \Rightarrow \text{max change in angle} &= \frac{\pi - \theta}{2} = 5E-3 \\
 &\approx \frac{\sin(\frac{\pi}{2} - \theta)}{2} \\
 &\approx \frac{\pi - \theta}{2}
 \end{aligned}$$



$$e_{EA} = e_{ij} n_i^{(A)} n_j^{(A)} = 10^{-3} = e_{11} \quad \text{(used } \underline{n}^{(A)} = (1 \ 0 \ 0)^T \text{)}$$

$$e_{EB} = e_{ij} n_i^{(B)} n_j^{(B)} = 2 \times 10^{-3} = \frac{1}{4} e_{11} + \frac{\sqrt{3}}{2} e_{12} + \frac{3}{4} e_{22} \quad \text{(used } \underline{n}^{(B)} = (\frac{1}{2} \ \frac{\sqrt{3}}{2} \ 0)^T \text{)}$$

$$e_{EC} = e_{ij} n_i^{(C)} n_j^{(C)} = 1.2 \times 10^{-3} = \frac{1}{4} e_{11} - \frac{\sqrt{3}}{2} e_{12} + \frac{3}{4} e_{22} \quad \text{(used } \underline{n}^{(C)} = (-\frac{1}{2} \ \frac{\sqrt{3}}{2} \ 0)^T \text{)}$$

$$\textcircled{1}, \textcircled{2}, \textcircled{3} \rightarrow e_{11} = 10^{-3}, \quad e_{22} = 1.8 \times 10^{-3}, \quad e_{12} = \frac{0.8}{\sqrt{3}} \times 10^{-3}$$

$$\sigma_{ij} = \frac{E}{1+\nu} \left[e_{ij} + \frac{\nu}{1-2\nu} e_{kk} \delta_{ij} \right] \rightarrow \sigma_{33} = 0 = \frac{E}{1+\nu} \left[e_{33} + \frac{\nu}{1-2\nu} e_{kk} \right]$$

$$\Rightarrow e_{33} = - \left(\frac{1-2\nu}{1-\nu} \right) \left(\frac{\nu}{1-2\nu} \right) (e_{11} + e_{22}) = - \frac{0.3}{0.7} (2.8 \times 10^{-3}) = -1.2 \times 10^{-3}$$

$$e = \begin{pmatrix} 1 & 0.8/\sqrt{3} & 0 \\ 0.8/\sqrt{3} & 1.8 & 0 \\ 0 & 0 & -1.2 \end{pmatrix} \times 10^{-3} \Rightarrow ((1-\lambda)(1.8-\lambda) - \frac{0.64}{3})(-1.2-\lambda) = 0$$

λ are p-strains

$$\lambda^2 - 2.8\lambda + \frac{119}{75} = 0, \quad \lambda = \frac{2.8 \pm 1.2220}{2} = (2.0110, 0.7890, -1.2) \times 10^{-3}$$

p-strains
max/min normal strains.

P-stresses: align coord system along p-axes of strain, which is same as p-axes of stress. Then use CL.

$$\sigma_{11} = \frac{2E5}{1.3} \left[2.0110 + \frac{0.3}{0.4} (1.6) \right] \times 10^{-3} = 4.94 E2 = \text{p-stress (1)} = N_{\max}$$

$$\sigma_{22} = \frac{2E5}{1.3} \left[0.7890 + \frac{0.3}{0.4} (1.6) \right] \times 10^{-3} = 3.06 E2 = \text{p-stress (2)} = N_{\min}$$

$$\sigma_{33} = 0 = \text{p-stress (3)}$$

$$S_{\max} = |4.94 - 0| \times E2 = 2.47 E2$$

Align coord system along S_{max} (say x_1') and normal to plane containing S_{max} (say x_2'). Then $\tau_{12}' = \text{max shear stress component}$. Hence,

$$\epsilon_{12}' = \frac{1+\nu}{E} \tau_{12}' = \frac{1+\nu}{E} S_{max} = 1.6055 \text{ E-3} = \text{max shear strain}$$

= max change in angle between two ^{perp} elements at P.

The two elements lie along x_1' & x_2' directions.

Alternately, you can directly use,

$$\text{max shear strain} = \left| \frac{\lambda(1) - \lambda(3)}{2} \right| = \left| \frac{2.011 + 1.2}{2} \right| \times 10^{-3}$$

Here the two elements corresponding to max shear strain (i.e. max change in angle) lie at 45° to $n(1)$, $n(2)$ (p-axes of strain)

