

P.1

$$\begin{aligned} \tau_{11,1} + \tau_{12,2} &= 0 \\ \tau_{12,1} + \tau_{22,2} &= 0 \end{aligned} \quad \left. \begin{array}{l} \text{when} \\ m_i \& f_i \\ \text{are zero} \end{array} \right\}$$

$$\Rightarrow \tau_{12} = \int \tau_{11,1} dx_2 + f(x_1) = -cx_2 + f(x_1) + K_1 \quad (\text{equate})$$

$$\Rightarrow \tau_{12} = \int -\tau_{22,2} dx_1 + g(x_2) = -bx_1 + g(x_2) + K_2$$

$$\therefore \tau_{12} = -bx_1 - cx_2 + K \quad (K_1 = K_2 = K) \quad \text{const}$$

On the face having unit normal in \mathbf{e}_1 dir.: (given $a=b=0$)

$$\tau_{13} = 0, \quad \tau_{12} = -cx_2 + K, \quad \tau_{11} = cx_1 + dx_2$$

In particular for $x_1 = -1$ face, $\tau_{11} = -c + dx_2$

On face having unit normal in \mathbf{e}_2 dir.: (for $a=b=0$, given)

$$\tau_{23} = 0, \quad \tau_{12} = -cx_2 + K, \quad \tau_{22} = 0$$

On top and bottom face, i.e., having \mathbf{e}_3 unit normal:

$$\tau_{13} = \tau_{23} = 0$$

Thus the b.c.'s imply that:

- (i) Plate unloaded on top & bottom face.
- (ii) On edges $x_1 = \text{const}$, plate has normal loading that varies linearly (with x_2) along the edge. This loading may change from tensile to compressive, depending on values of c, d .

On edges $x_1 = \text{const}$, plate has linearly varying ^(in-plane) shear loading that changes direction at $x_2 = K/c$ (i.e., it is zero at $x_2 = K/c$). If $c=0$ then in-plane shear loading is const. ($=K$)

- (iii) On edges $x_2 = \text{const}$ there is only a constant in-plane shear loading whose magnitude depends c, K & the location of the edge (i.e., x_2 [const] value).

P.2

When m_i & f_i are zero, the equil. eqns are,

$$\begin{aligned} \tau_{11,1} + \tau_{12,2} &= 0 \\ \tau_{12,1} + \tau_{22,2} &= 0 \end{aligned} \quad \Rightarrow \exists F(x_1, x_2) : \quad \begin{array}{l} \tau_{11} = F_{,2} \& \tau_{12} = -F_{,1} \\ \tau_{12} = -G_{,2} \& \tau_{22} = G_{,1} \end{array} \quad \xrightarrow{\text{(This is a standard result from Calculus)}}$$

$$\therefore \tau_{12} = -F_{,1} = -G_{,2} \Rightarrow \exists \phi(x_1, x_2) : F = \phi_{,2} \& G = \phi_{,1}$$

$$\therefore \tau_{11} = F_{,2} = \phi_{,22}, \quad \tau_{22} = G_{,1} = \phi_{,11}, \quad \tau_{12} = -F_{,1} = -G_{,2} = -\phi_{,12} \quad \blacktriangleleft$$

(2)

$$\begin{aligned}
 n_1(1) n_1(2) = 0 &\Rightarrow \frac{1}{2} n_2(2) + \sqrt{\frac{1}{2}} n_3(2) = 0 \quad (\text{Note: } n_1(2) = 0 \text{ from } \phi) \\
 n_1(2) n_1(3) = 1 &\Rightarrow n_2^2(2) + n_3^2(2) = 1
 \end{aligned}
 \left. \begin{array}{l} \Rightarrow n_3(2) = \pm \sqrt{\frac{1}{3}}, n_2(2) = \mp \sqrt{\frac{2}{3}}, \text{ so } n_1(2) \Rightarrow (0, -\sqrt{\frac{2}{3}}, \pm \sqrt{\frac{1}{3}}) \\ n_1(3) = \in_{ijk} n_j(1) n_k(2) \Rightarrow n_1(3) = n_2(1) n_3(2) - n_3(1) n_2(2) = \frac{1}{2} \left(\sqrt{\frac{1}{3}} \right) - \sqrt{\frac{1}{2}} \left(-\sqrt{\frac{2}{3}} \right) = \sqrt{\frac{1}{12}} + \sqrt{\frac{1}{3}} \end{array} \right\}$$

$$\begin{aligned}
 n_2(3) &= n_2(1) n_1(2) - n_1(1) n_2(2) = -\frac{1}{2} \sqrt{\frac{1}{3}} = -\sqrt{\frac{1}{12}} \\
 n_3(3) &= n_1(1) n_2(2) - n_2(1) n_1(2) = \frac{1}{2} \left(-\sqrt{\frac{2}{3}} \right) = -\sqrt{\frac{1}{6}}
 \end{aligned}$$

$$\therefore n_i(3) \Rightarrow \left(\sqrt{\frac{1}{12}} + \sqrt{\frac{1}{3}}, -\sqrt{\frac{1}{12}}, -\sqrt{\frac{1}{6}} \right)$$

Let x'_i be the principal axes system.

$$\text{Then } \tau'_{11} = \tau(1) = 1, \tau'_{22} = \tau(2) = 4, \tau'_{33} = \tau(3) = -2, \tau'_{ij} = 0 \text{ for } i \neq j$$

$$\text{Now } \tau'_{ij} = a_{mi} a_{nj} \tau_{mn} \text{ where } a_{mn} \Rightarrow \begin{pmatrix} n_1(1) & n_2(1) & n_3(1) \\ n_1(2) & n_2(2) & n_3(2) \\ n_1(3) & n_2(3) & n_3(3) \end{pmatrix}$$

$$\therefore \tau'_{11} = n_1^2(1) \tau(1) + n_1^2(2) \tau(2) + n_1^2(3) \tau(3) = -1.25$$

$$\tau'_{22} = n_2^2(1) \tau(1) + n_2^2(2) \tau(2) + n_2^2(3) \tau(3) = 2.75$$

$$\tau'_{33} = n_3^2(1) \tau(1) + n_3^2(2) \tau(2) + n_3^2(3) \tau(3) = 1.5$$

$$\tau'_{12} = n_1(1) n_2(1) \tau(1) + n_1(2) n_2(2) \tau(2) + n_1(3) n_2(3) \tau(3) = 0.75$$

$$\tau'_{13} = n_1(1) n_3(1) \tau(1) + n_1(2) n_3(2) \tau(2) + n_1(3) n_3(3) \tau(3) = 1.061$$

$$\tau'_{23} = n_2(1) n_3(1) \tau(1) + n_2(2) n_3(2) \tau(2) + n_2(3) n_3(3) \tau(3) = -1.768$$

$$\Rightarrow \tau'_{ij} \Rightarrow \begin{pmatrix} -1.25 & 0.75 & 1.061 \\ 0.75 & 2.75 & -1.768 \\ 1.061 & -1.768 & 1.5 \end{pmatrix}$$

$$\textcircled{P4}(a)(i) \phi = 2x_1 + x_2 + x_3 - 6 = 0, n_i = \frac{\phi_i}{(\phi_j \phi_k)^{1/2}} = \frac{2e_1 + e_2 + e_3}{\sqrt{6}} \Rightarrow \left(\frac{2}{\sqrt{6}}, \frac{1}{\sqrt{6}}, \frac{1}{\sqrt{6}} \right)$$

$$\tau_i = \tau_{ij} n_j = \tau_{ij} n_j \Rightarrow \left(-\alpha x_2 \frac{1}{\sqrt{6}}, \alpha x_1 \frac{1}{\sqrt{6}}, [-\alpha x_2 \frac{2}{\sqrt{6}} + \alpha x_1 \frac{1}{\sqrt{6}}] \right)$$

$$\therefore \tau_i @ (1, 2, 2) \Rightarrow \left(-\frac{2a}{\sqrt{6}}, \frac{a}{\sqrt{6}}, -\frac{3a}{\sqrt{6}} \right) \blacktriangleleft$$

$$N = \tau_{ij} n_i n_j = -a \blacktriangleleft$$

$$S = (\tau_i \tau_i - N^2)^{1/2} = \frac{2}{\sqrt{3}} a \blacktriangleleft$$

$$\text{(ii)} \phi = x_1 x_2 - 9 = 0, n_j = \frac{\phi_j}{(\phi_k \phi_k)^{1/2}} = \frac{2x_j}{2(x_1^2 + x_2^2 + x_3^2)^{1/2}}$$

$$\therefore \tau_i = \tau_{ij} n_j \Rightarrow \left(\frac{-\alpha x_2 x_3}{(x_1^2 + x_2^2 + x_3^2)^{1/2}}, \frac{\alpha x_1 x_3}{(x_1^2 + x_2^2 + x_3^2)^{1/2}}, 0 \right)$$

$$\tau_i @ (1, 2, 2) \Rightarrow \left(-\frac{4a}{3}, \frac{2a}{3}, 0 \right) \blacktriangleleft$$

$$N = \tau_{ij} n_i n_j = 0 \blacktriangleleft$$

$$S = (\tau_i \tau_i - N^2)^{1/2} = \frac{2\sqrt{5}}{3} a \blacktriangleleft$$

P4(b) $\sigma_{ij} \Rightarrow \begin{pmatrix} 0 & 0 & -2a \\ 0 & 0 & a \\ -2a & a & 0 \end{pmatrix} \Rightarrow \begin{vmatrix} -\tau & 0 & -2a \\ 0 & -\tau & a \\ -2a & a & -\tau \end{vmatrix} = -\tau(\tau^2 - a^2) - 2a(-2a\tau) = 0 \Rightarrow \tau^3 - 5\tau a^2 = 0 \Rightarrow \tau(1) = a\sqrt{5}, \tau(2) = 0, \tau(3) = -a\sqrt{5}$

$$\left. \begin{array}{l} -\tau n_1 - 2an_3 = 0 \\ -\tau n_2 + an_3 = 0 \end{array} \right\} \text{so, } n_i \Rightarrow \left(-\frac{2a}{\tau} n_3, \frac{a}{\tau} n_3, n_3 \right) \text{ when } \tau \neq 0$$

$$n_1^2 + n_2^2 + n_3^2 = 1 \Rightarrow n_3 = \frac{1}{\sqrt{\left(\frac{4a^2}{\tau^2} + \frac{a^2}{\tau^2} + 1\right)^{1/2}}} \text{ when } \tau \neq 0$$

$$\text{So, } n_i(1) \Rightarrow \left(\pm \frac{2}{\sqrt{10}}, \pm \frac{1}{\sqrt{10}}, \pm \frac{1}{\sqrt{2}} \right) \Rightarrow n_i(3) \Rightarrow \left(\pm \frac{2}{\sqrt{10}}, \mp \frac{1}{\sqrt{10}}, \pm \frac{1}{\sqrt{2}} \right)$$

When $\tau = 0$, $n_3 = 0$, we must use 3rd eqn, i.e., $-2an_1 + an_2 - \tau n_3 = 0$ or else use $n_i(2) = \epsilon_{ijk} n_j(1) n_k(3)$.

$$\text{Thus } n_i(2) \Rightarrow (n_1, 2n_1, 0)$$

$$n_i(2) n_i(2) = 1 \Rightarrow n_1 = \frac{1}{\sqrt{5}} \Rightarrow n_i(2) \Rightarrow \left(\pm \frac{1}{\sqrt{5}}, \pm \frac{2}{\sqrt{5}}, 0 \right)$$

$$S(1) = \left| \frac{\tau(2) - \tau(3)}{2} \right|, S(2) = \left| \frac{\tau(3) - \tau(1)}{2} \right|, S(3) = \left| \frac{\tau(1) - \tau(2)}{2} \right|$$

So max shearing stress is $S = S(2) = a\sqrt{5}$

P.5

Principal stresses: $|\sigma_{ij} - \tau \delta_{ij}| = 0 \Rightarrow \tau^3 - 6\tau^2 - 4\tau + 24 = 0 \Rightarrow (\tau - 6)(\tau^2 - 4) = 0 \Rightarrow \tau(1) = 6, \tau(2) = 2, \tau(3) = -2$

(a) Principal Deviator Stresses

$$\hat{\tau}(1) = \tau(1) - \frac{1}{3}\tau_{kk} = \tau(1) - \frac{6}{3} = 4$$

$$\hat{\tau}(2) = \tau(2) - \frac{1}{3}\tau_{kk} = 0$$

$$\hat{\tau}(3) = \tau(3) - \frac{1}{3}\tau_{kk} = -4$$

(b) Let x'_i = principal directions & $n'_i \Rightarrow \left(\frac{1}{\sqrt{3}}, \frac{1}{\sqrt{3}}, \frac{1}{\sqrt{3}} \right)$ is the octahedral plane
In general (for any n'_i), $N = \tau(1)n'_1^2 + \tau(2)n'_2^2 + \tau(3)n'_3^2$

$$\therefore N_{oct} = \frac{1}{3}(\tau(1) + \tau(2) + \tau(3)) = \frac{1}{3}\tau_{ii} = 2$$

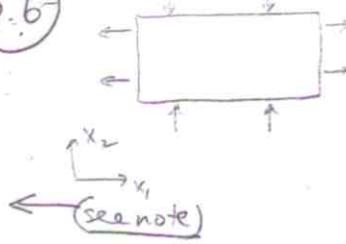
$$(\tau'_i)_{oct} \Rightarrow (\tau(1)n'_1, \tau(2)n'_2, \tau(3)n'_3) \Rightarrow \frac{1}{\sqrt{3}}(6, 2, -2)$$

$$\therefore S_{oct} = \left[(\tau'_i \tau'_i)_{oct} - N_{oct}^2 \right]^{1/2} = \left(\frac{44}{3} - 4 \right)^{1/2} = \left(\frac{32}{3} \right)^{1/2} = \frac{4}{3}\sqrt{6}$$

(c) $S_{max} = \left| \frac{\tau(3) - \tau(1)}{2} \right| = 4$

4

P.6



B.C.'s : edges $x_1 = \text{const}$: $\tau_{11} = \tau_{11}^*$, $\tau_{12} = 0$, $\tau_{13} = 0$
edges $x_2 = \text{const}$: $\tau_{22} = -\tau_{22}^*$, $\tau_{12} = 0$, $\tau_{23} = 0$
face $x_3 = \text{const}$: $\tau_{13} = 0$

Plane State

of Stress

Equil: $\tau_{ij,j} = 0 \Rightarrow \begin{cases} \tau_{11,1} + \tau_{12,2} = 0 \\ \tau_{12,1} + \tau_{22,2} = 0 \end{cases} \left\{ \begin{array}{l} \tau_{11} = \tau_{11}^*, \tau_{22} = -\tau_{22}^* \\ \tau_{12} = 0 \end{array} \right. \text{ is soln. that satisfies equil & B.C.'s} \right\}$

Thus $\tau_{ij} \Rightarrow \begin{pmatrix} \tau_{11}^* & 0 & 0 \\ 0 & -\tau_{22}^* & 0 \\ 0 & 0 & 0 \end{pmatrix} \rightarrow \text{principal stresses are } \sigma(1) = \tau_{11}^*, \sigma(2) = -\tau_{22}^*, \text{ & p-axes are } n_i(1) \Rightarrow (1, 0, 0), n_i(2) \Rightarrow (0, 1, 0), n_i(3) \Rightarrow (0, 0, 1) \right\}$

Thus max shearing stress is $\frac{1}{2} |\sigma(1) - \sigma(2)| = \frac{\tau_{11}^* + \tau_{22}^*}{2}$ which acts on plane having unit normal $(\pm \frac{1}{\sqrt{2}}, \pm \frac{1}{\sqrt{2}}, 0)$. But this is precisely the plane containing the flow so the shearing stress in the flow plane is also the max shearing stress.

∴ at $S_{\max} > \tau_{cr}$ the plate will fail.

so $\boxed{\tau_{11}^* + \tau_{22}^* > 2\tau_{cr}}$ → is failure condition

P.7

$i=1$ $\tau_{11,1} + \tau_{12,2} + \tau_{13,3} + f_1 = 0$

$\uparrow e_3, \downarrow e_2 = e'_3$
 $e_1, e_2 = e'_1$
 $r\theta = e'_2$
 $r\phi = e'_1$
 $e_r = e'_1$

$a_{ij} = \begin{pmatrix} c\theta & s\theta & 0 \\ -s\theta & c\theta & 0 \\ 0 & 0 & 1 \end{pmatrix}$

let, $\tau_{ij} = \tau_{ij}[x_1, x_2, x_3] \equiv \tau_{ij}[r, \theta, z]$
 $\tau'_{ij} = \tau'_{ij}[r, \theta, z] \equiv \tau'_{ij}[x_1, x_2, x_3]$
 $f_i = f_i[x_1, x_2, x_3] \equiv f_i[r, \theta, z]$

Now, $\tau_{11,1} = \frac{\partial \tau_{11}}{\partial x_1} = \frac{\partial \tau_{11}}{\partial r} \frac{\partial r}{\partial x_1} + \frac{\partial \tau_{11}}{\partial \theta} \frac{\partial \theta}{\partial x_1} + \frac{\partial \tau_{11}}{\partial z} \frac{\partial z}{\partial x_1}$

$c\theta$
 $-s\theta$
 $\tau_{12,2} = \frac{\partial \tau_{12}}{\partial x_2} = \frac{\partial \tau_{12}}{\partial r} \frac{\partial r}{\partial x_2} + \frac{\partial \tau_{12}}{\partial \theta} \frac{\partial \theta}{\partial x_2} + \frac{\partial \tau_{12}}{\partial z} \frac{\partial z}{\partial x_2}$

$s\theta$
 $\frac{c\theta}{r}$
 $\tau_{13,3} = \frac{\partial \tau_{13}}{\partial x_3} = \frac{\partial \tau_{13}}{\partial r} \frac{\partial r}{\partial x_3} + \frac{\partial \tau_{13}}{\partial \theta} \frac{\partial \theta}{\partial x_3} + \frac{\partial \tau_{13}}{\partial z} \frac{\partial z}{\partial x_3}$

$\therefore \tau_{11,1} = c\theta \tau_{11,r} - \frac{s\theta}{r} \tau_{11,\theta} = \frac{c^3 \theta \tau_{11}'}{r} - 2c^2 \theta s\theta \tau_{12,r}' + c\theta s^2 \theta \tau_{22,r}' - \frac{s\theta}{r} (-2c\theta s\theta \tau_{11}' + c^2 \theta \tau_{12}') + 2[s^2 \theta - c^2 \theta] \tau_{12}' - 2c\theta s\theta \tau_{12,\theta}' + 2s\theta c\theta \tau_{22,\theta}' + s^2 \theta \tau_{22,\theta}'$

Now $\tau_{ij} = a_{mi} a_{nj} \tau'_{mn}$

$\tau_{11} = c^2 \theta \tau'_{11} - 2c\theta s\theta \tau'_{12} + s^2 \theta \tau'_{22}$
 $\tau_{12} = c\theta s\theta \tau'_{11} + c^2 \theta \tau'_{12} - s^2 \theta \tau'_{21} - s\theta c\theta \tau'_{22}$
 $= c\theta s\theta (\tau'_{11} - \tau'_{22}) + (c^2 \theta - s^2 \theta) \tau'_{12}$

$\tau_{13} = c\theta \tau'_{13} - s\theta \tau'_{23}$
 $r^2 = x_1^2 + x_2^2 \Rightarrow 2r \frac{\partial r}{\partial x_1} = 2x_1 \Rightarrow \frac{\partial r}{\partial x_1} = x_1/r = \frac{r \cos \theta}{r}$

$\tan \theta = \frac{x_2}{x_1} \Rightarrow (1 + t^2 \theta) \frac{\partial \theta}{\partial x_1} = -\frac{x_2}{x_1^2} = -\frac{t \theta}{x_1}$
 $\therefore \frac{\partial \theta}{\partial x_1} = -\frac{c^2 \theta + \theta}{x_1} = -\frac{c\theta s\theta}{r c\theta} = -\frac{s\theta}{r}$

$\therefore \frac{\partial \theta}{\partial x_2} = \frac{c^2 \theta}{x_1} = \frac{c^2 \theta}{r c\theta} = \frac{c\theta}{r}$

$$\tau_{12,2} = S\theta C\theta (\tilde{\tau}_{11,r}' - \tilde{\tau}_{22,r}') + S\theta (C^2\theta - S^2\theta) \tilde{\tau}_{12,r}' + \frac{C\theta}{r} \left\{ [C^2\theta - S^2\theta] (\tilde{\tau}_{11}' - \tilde{\tau}_{22}') + C\theta S\theta (\tilde{\tau}_{11,\theta}' - \tilde{\tau}_{22,\theta}')$$

$$+ [-2C\theta S\theta - 2S\theta C\theta] \tilde{\tau}_{12}' + [C^2\theta - S^2\theta] \tilde{\tau}_{12,\theta}' \right\}$$

Now put $\tilde{\tau}_{11}' = \tilde{\tau}_{rr}$, $\tilde{\tau}_{22}' = \tilde{\tau}_{\theta\theta}$, $\tilde{\tau}_{12}' = \tilde{\tau}_{r\theta}$, $\tilde{\tau}_{13}' = \tilde{\tau}_{rz}$, $\tilde{\tau}_{23}' = \tilde{\tau}_{\theta z}$

$$\therefore \tilde{\tau}_{11,1} + \tilde{\tau}_{12,2} + \tilde{\tau}_{13,3} + f_1$$

$$= \tilde{\tau}_{rr,r} C\theta - \tilde{\tau}_{r\theta,r} S\theta + \frac{\tilde{\tau}_{rr} - \tilde{\tau}_{\theta\theta}}{r} C\theta - 2\frac{\tilde{\tau}_{r\theta}}{r} S\theta - \frac{\tilde{\tau}_{\theta\theta,\theta}}{r} S\theta + \frac{\tilde{\tau}_{r\theta,\theta}}{r} C\theta + \tilde{\tau}_{rz,z} C\theta - \tilde{\tau}_{\theta z,z} S\theta$$

$$+ f_r C\theta - f_\theta S\theta = 0$$

∴ $C\theta \left(\tilde{\tau}_{rr,r} + \frac{\tilde{\tau}_{rr} - \tilde{\tau}_{\theta\theta}}{r} + \frac{\tilde{\tau}_{r\theta,\theta}}{r} + \tilde{\tau}_{rz,z} + f_r \right) - S\theta \left(\tilde{\tau}_{r\theta,r} + 2\frac{\tilde{\tau}_{r\theta}}{r} + \frac{\tilde{\tau}_{\theta\theta,\theta}}{r} + \tilde{\tau}_{\theta z,z} + f_\theta \right) = 0 \quad \rightarrow ①$

Extension of problem:

$$\underline{i=2} \quad \tilde{\tau}_{12,1} + \tilde{\tau}_{22,2} + \tilde{\tau}_{23,3} = 0$$

$$\tilde{\tau}_{22} = S^2\theta \tilde{\tau}_{11}' + 2S\theta C\theta \tilde{\tau}_{12}' + C^2\theta \tilde{\tau}_{22}' ; \quad \tilde{\tau}_{23} = S\theta \tilde{\tau}_{13}' + C\theta \tilde{\tau}_{23}'$$

$$\tilde{\tau}_{12,1} = C\theta \tilde{\tau}_{12,r} - \frac{S\theta}{r} \tilde{\tau}_{12,\theta} ; \quad \tilde{\tau}_{22,2} = S\theta \tilde{\tau}_{22,r} + \frac{C\theta}{r} \tilde{\tau}_{22,\theta} ; \quad \tilde{\tau}_{23,3} = \tilde{\tau}_{23,z}$$

$$\tilde{\tau}_{12,1} = C^2\theta S\theta (\tilde{\tau}_{11,r}' - \tilde{\tau}_{22,r}') + (C^3\theta - C\theta S^2\theta) \tilde{\tau}_{12,r}'$$

$$- \frac{S\theta}{r} \left\{ [C^2\theta - S^2\theta] (\tilde{\tau}_{11}' - \tilde{\tau}_{22}') + C\theta S\theta (\tilde{\tau}_{11,\theta}' - \tilde{\tau}_{22,\theta}') + [-4C\theta S\theta] \tilde{\tau}_{12}' + [C^2\theta - S^2\theta] \tilde{\tau}_{12,\theta}' \right\}$$

$$\tilde{\tau}_{22,2} = S^3\theta \tilde{\tau}_{11,r}' + 2S^2\theta C\theta \tilde{\tau}_{12,r}' + S\theta C^2\theta \tilde{\tau}_{22,r}'$$

$$+ \frac{C\theta}{r} \left\{ 2S\theta C\theta \tilde{\tau}_{11}' + S^2\theta \tilde{\tau}_{11,\theta}' + 2[C^2\theta - S^2\theta] \tilde{\tau}_{12}' + 2S\theta C\theta \tilde{\tau}_{12,\theta}' - 2C\theta S\theta \tilde{\tau}_{22}' + C^2\theta \tilde{\tau}_{22,\theta}' \right\}$$

$$\therefore \tilde{\tau}_{12,1} + \tilde{\tau}_{22,2} + \tilde{\tau}_{23,3} + f_2$$

$$= \tilde{\tau}_{rr,r} S\theta + \tilde{\tau}_{r\theta,r} C\theta + \frac{\tilde{\tau}_{rr} - \tilde{\tau}_{\theta\theta}}{r} S\theta + 2\frac{\tilde{\tau}_{r\theta}}{r} C\theta + \frac{\tilde{\tau}_{\theta\theta,\theta}}{r} S\theta + \frac{\tilde{\tau}_{\theta\theta,\theta}}{r} C\theta + f_r S\theta + f_\theta C\theta$$

$$\therefore S\theta \left(\tilde{\tau}_{rr,r} + \frac{\tilde{\tau}_{rr} - \tilde{\tau}_{\theta\theta}}{r} + \frac{\tilde{\tau}_{r\theta,\theta}}{r} + \tilde{\tau}_{rz,z} + f_r \right) + C\theta \left(\tilde{\tau}_{r\theta,r} + 2\frac{\tilde{\tau}_{r\theta}}{r} + \frac{\tilde{\tau}_{\theta\theta,\theta}}{r} + \tilde{\tau}_{\theta z,z} + f_\theta \right) = 0 \quad \rightarrow ②$$

From ① & ② we get,

$$\begin{aligned} \tilde{\tau}_{rr,r} + \frac{\tilde{\tau}_{rr} - \tilde{\tau}_{\theta\theta}}{r} + \frac{\tilde{\tau}_{r\theta,\theta}}{r} + \tilde{\tau}_{rz,z} + f_r &= 0 \\ \tilde{\tau}_{r\theta,r} + 2\frac{\tilde{\tau}_{r\theta}}{r} + \frac{\tilde{\tau}_{\theta\theta,\theta}}{r} + \tilde{\tau}_{\theta z,z} + f_\theta &= 0 \end{aligned} \quad \left. \begin{array}{l} \text{two of the} \\ \text{eqns in cylindrical coords} \end{array} \right\}$$

Q.8

$$\tau_{11,1} + \tau_{12,2} = 0 \Rightarrow \tau_{12,2} = -\frac{q}{2I} [2x_1 x_2] \Rightarrow \tau_{12} = -\frac{q}{2I} x_1 x_2^2 + f(x_1) + c,$$

$$\tau_{12,1} + \tau_{22,2} = 0 \Rightarrow \tau_{12,1} = -\frac{q}{2I} [x_2^2 - c^2] \Rightarrow \tau_{12} = -\frac{q}{2I} [x_1 x_2^2 - c^2 x_1] + g(x_2) + c_2$$

$$\Rightarrow f(x_1) = \frac{q}{2I} c^2 x_1, \quad g(x_2) = 0, \quad c_1 = c_2 = k$$

$$\therefore \boxed{\tau_{12} = \frac{q}{2I} x_1 (c^2 - x_2^2) + k} \quad \text{(const)}$$

B.C's Let applied load @ boundary be τ_{11}^* , τ_{22}^* , τ_{12}^* .

$$\text{at } x_1=0, \quad \tau_{11}^* = \frac{q}{I} x_2 \left(\frac{c^2}{5} - \frac{x_2^2}{3} \right), \quad \tau_{12}^* = k, \quad \tau_{13}^* = 0$$

$$\text{at } x_1=2b, \quad \tau_{11}^* = \frac{q}{I} x_2 \left(2b^2 - \frac{x_2^2}{3} + \frac{c^2}{5} \right), \quad \tau_{12}^* = \frac{q}{I} b (c^2 - x_2^2) + k, \quad \tau_{13}^* = 0$$

$$\text{at } x_2=+c, \quad \tau_{22}^* = 0, \quad \tau_{12}^* = k, \quad \tau_{23}^* = 0$$

$$\text{at } x_2=-c, \quad \tau_{22}^* = \frac{q}{2I} \frac{4c^3}{3} = q, \quad \tau_{12}^* = k, \quad \tau_{23}^* = 0$$

$$\text{at } x_3=0, t, \quad \tau_{33}^* = \tau_{31}^* = \tau_{32}^* = 0 \quad (\text{ie top and bottom faces unloaded})$$

Q.9(a) For a plane equally inclined to the 3-principal axes we have the stress vector = τ_{oct} .

Now, $\tau_1 = \tau(1)n_1$; $\tau_2 = \tau(2)n_2$; $\tau(3) = \tau_3 n_3$. (by letting the M_i -axes coincide with the principal axes and using the Cauchy equation).

Thus the stress tensor has the form :- $\tau_{ij} = \begin{pmatrix} \tau(1) & 0 & 0 \\ 0 & \tau(2) & 0 \\ 0 & 0 & \tau(3) \end{pmatrix}$

\therefore for the octahedral plane,

$$n_1 = n_2 = n_3 = \pm \frac{1}{\sqrt{3}}$$

$$\therefore \tau_i \sigma_i = \tau_1^2 + \tau_2^2 + \tau_3^2 = \tau(1)^2 n_1^2 + \tau(2)^2 n_2^2 + \tau(3)^2 n_3^2.$$

$$\therefore \tau_{\text{oct}} = \sqrt{\frac{1}{3} (\tau(1)^2 + \tau(2)^2 + \tau(3)^2)}.$$

$$(b) \quad \cancel{N^2} \quad N = \tau_i n_i = \epsilon_{ijk} n_j n_k$$

Now the shearing components of the stress tensor are zero, from our choice of coordinate axes

$$\therefore N = \cancel{\tau_{ij} n_i n_j} + \tau_{22} n_2 n_2 + \tau_{33} n_3 n_3$$

$$= \tau_{11} \frac{1}{3} + \tau_{22} \frac{1}{3} + \tau_{33} \frac{1}{3} \quad (\text{since } n_1 = n_2 = n_3 = \pm \frac{1}{\sqrt{3}})$$

$$\therefore N = \frac{1}{3} (\tau_{kk}) = \frac{1}{3} \textcircled{H}$$

$$\therefore N = \frac{1}{3} \textcircled{H}$$

$$(c) \quad S^2 = \sigma_i \tau_i - N^2$$

$$= \tau(1)^2 n_1^2 + \tau(2)^2 n_2^2 + \tau(3)^2 n_3^2 - (\tau(1)n_1^2 + \tau(2)n_2^2 + \tau(3)n_3^2)^2$$

$$= \tau_{11}^2 n_1^2 + \tau_{22}^2 n_2^2 + \tau_{33}^2 n_3^2 - (\tau_{11} n_1^2 + \tau_{22} n_2^2 + \tau_{33} n_3^2)^2$$

$$\therefore S^2_{\text{act}} = \frac{1}{3} (\tau_{11}^2 + \tau_{22}^2 + \tau_{33}^2) - \frac{1}{3} [\tau_{11}^2 + \tau_{22}^2 + \tau_{33}^2 + 2\tau_{11}\tau_{22} + 2\tau_{11}\tau_{33} + 2\tau_{22}\tau_{33}]$$

$$(I) \leftarrow \therefore S_{\text{act}} = \left[\frac{1}{3} \left(\frac{2}{3} \tau_{11}^2 + \frac{2}{3} \tau_{22}^2 + \frac{2}{3} \tau_{33}^2 - \frac{2}{3} \tau_{11}\tau_{22} - \frac{2}{3} \tau_{22}\tau_{33} - \frac{2}{3} \tau_{11}\tau_{33} \right) \right]$$

Now let us examine the expression $(-\frac{2}{3} \hat{\Phi})^{1/2}$

$$\hat{\sigma}_{ij} = \begin{pmatrix} \left(\frac{2}{3} \tau_{11} - \frac{1}{3} \tau_{22} - \frac{1}{3} \tau_{33}\right) & 0 & 0 \\ 0 & \left(\frac{2}{3} \tau_{22} - \frac{1}{3} \tau_{11} - \frac{1}{3} \tau_{33}\right) & 0 \\ 0 & 0 & \left(\frac{2}{3} \tau_{33} - \frac{1}{3} \tau_{22} - \frac{1}{3} \tau_{11}\right) \end{pmatrix}$$

$$\therefore -\frac{2}{3} \hat{\Phi} = -\frac{2}{3} \left[-\frac{2}{9} \tau_{11}^2 - \frac{2}{9} \tau_{22}^2 + \frac{1}{9} \tau_{33}^2 + \frac{5}{9} \tau_{11} \tau_{22} - \frac{1}{9} \tau_{11} \tau_{33} - \frac{1}{9} \tau_{22} \tau_{33} \right. \\ \left. + -\frac{2}{9} \tau_{11}^2 - \frac{2}{9} \tau_{33}^2 + \frac{1}{9} \tau_{22}^2 + \frac{5}{9} \tau_{11} \tau_{33} - \frac{1}{9} \tau_{11} \tau_{22} - \frac{1}{9} \tau_{33} \tau_{22} \right. \\ \left. - \frac{2}{9} \tau_{22}^2 - \frac{2}{9} \tau_{33}^2 + \frac{1}{9} \tau_{11}^2 + \frac{5}{9} \tau_{22} \tau_{33} - \frac{1}{9} \tau_{22} \tau_{11} - \frac{1}{9} \tau_{33} \tau_{11} \right] \\ = -\frac{2}{3} \left[-\frac{2}{9} \tau_{11}^2 - \frac{2}{9} \tau_{22}^2 - \frac{2}{9} \tau_{33}^2 + \frac{3}{9} \tau_{11} \tau_{22} + \frac{3}{9} \tau_{11} \tau_{33} + \frac{3}{9} \tau_{22} \tau_{33} \right]$$

$$(II) \leftarrow \therefore \left[-\frac{2}{3} \hat{\Phi} \right]^{1/2} = \left\{ \frac{1}{3} \left[\frac{2}{3} \tau_{11}^2 + \frac{2}{3} \tau_{22}^2 + \frac{2}{3} \tau_{33}^2 - \frac{2}{3} \tau_{11} \tau_{22} - \frac{2}{3} \tau_{11} \tau_{33} - \frac{2}{3} \tau_{22} \tau_{33} \right] \right\}^{1/2}$$

$\therefore (I) = (II) \rightarrow \text{hence proved.}$

(8)

P10

The state of stress satisfying all equilibrium eqns and b.c.s is
 $\tau_{11} = P_{\max}/A$, other components are zero.

Case I : $S_{\max} = \left| \frac{\sigma_{11}}{2} \right| = \frac{\tau_{11}}{2} = \frac{P_{\max}}{2A} \leq 300 \times 10^3 \Rightarrow P_{\max} = 10^5 A = 10N$.

also $\tau_{11} = \frac{P_{\max}}{A} \leq 10^5$ } i.e., tensile failure along plane \perp or \parallel x_1 axis.

Case II : $S_{\max} = \left| \frac{\sigma_{11}}{2} \right| = \frac{\tau_{11}}{2} = \frac{P_{\max}}{2A} \leq 450 \times 10^3 \Rightarrow$

also $\tau_{11} = \frac{P_{\max}}{A} \leq 10^6$ } $P_{\max} = 0.9 \times 10^6 A = 90N$
 ie shear failure along plane inclined at 45° to x_1 axis (\because p-axes for σ_{12}, σ_{13})
 are any two orthogonal axes in x_2-x_3 plane.

(11)

$$\begin{vmatrix} (80-\sigma) & 45 & 0 \\ 45 & (-35-\sigma) & 0 \\ 0 & 0 & (50-\sigma) \end{vmatrix} = 0$$

$$-\sigma^3 + 5\sigma^2 + 70756 + 241250 = 0$$

$$\sigma_1 = +95.52 \text{ MPa}$$

$$\sigma_2 = -50.52 \text{ MPa}$$

$$\sigma_3 = -50.0 \text{ MPa.}$$

$$\tau_{max} = \pm \left(\frac{\sigma_1 - \sigma_2}{2} \right)$$

$$= \pm 73.02 \text{ MPa.}$$

$$S_{oct} = \frac{\sqrt{2}}{3} (l_1^2 - 3l_2)^{1/2}$$

$$= 68.72 \text{ MPa.}$$

(12)

$$\begin{vmatrix} (-150-\sigma) & -40 & 50 \\ -40 & -\sigma & 0 \\ 50 & 0 & 80-\sigma \end{vmatrix} = 0$$

$$-\sigma^3 - 70\sigma^2 + 16100\sigma + 128000 = 0$$

$$\sigma_1 = +91.18 \text{ MPa}$$

$$\sigma_2 = +8.28 \text{ MPa}$$

$$\sigma_3 = -169.46 \text{ MPa.}$$

$$\tau_{max} = 130.32 \text{ MPa.}$$

(13)

$$\sum F_y = 0$$

$$40 \times 5 + \sigma_{xy} \times 12 = 80 \times 13 \times \sin \theta + 60 \times 13 \times \cos \theta$$

$$\sigma_{xy} = 76.66 \text{ MPa.}$$

$$\sum F_x = 0$$

$$80 \times 13 \times \cos \theta = 60 \times 13 \times \sin \theta + \sigma_{xy} \times 5 + \sigma_{xx} \times 12$$

$$\sigma_{xx} = 23.03 \text{ MPa.}$$

